



SG2224
Applied CFD
17 April 2012

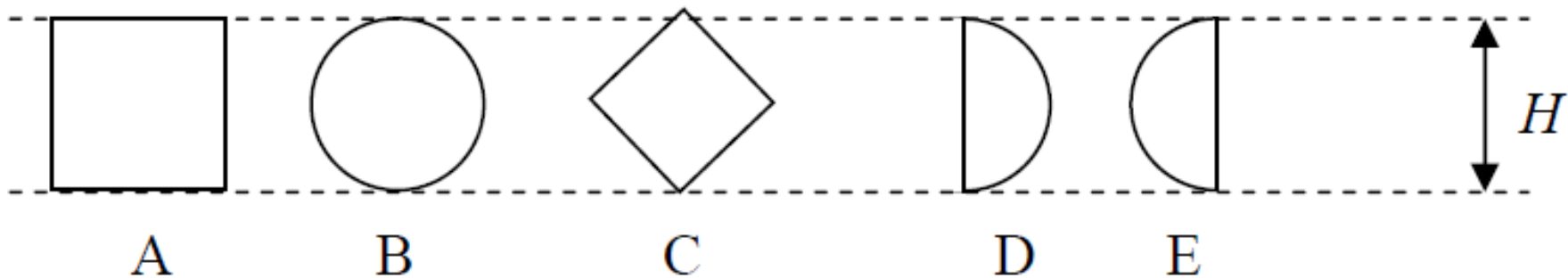
Today



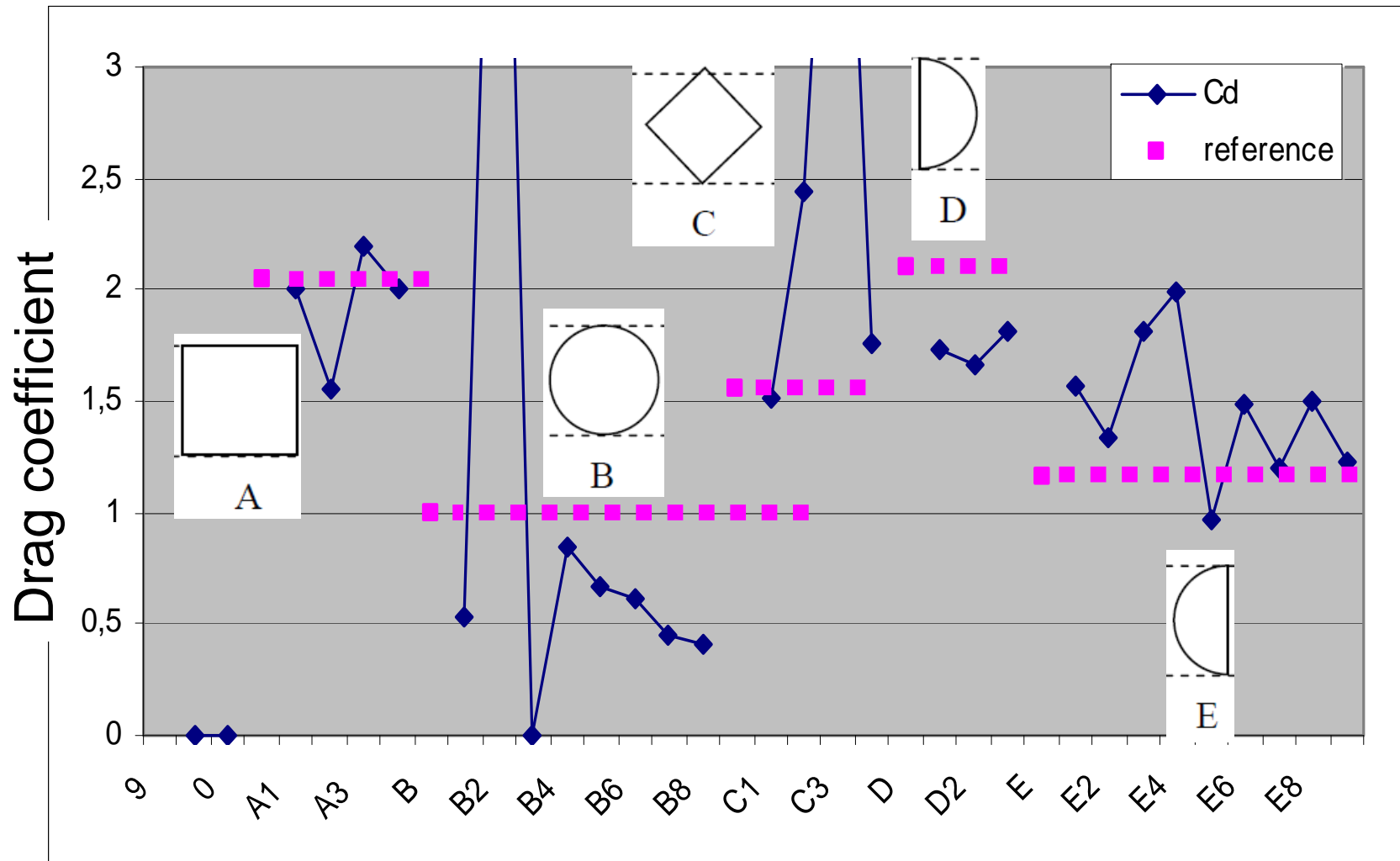
- Individual task
- Project information
 - Time to start project work
 - Contact your project coordinator (appointment)
 - Use bilda – project group discussion forum
- Lectures:
 - boundary conditions
 - quality and best practice

Individual task

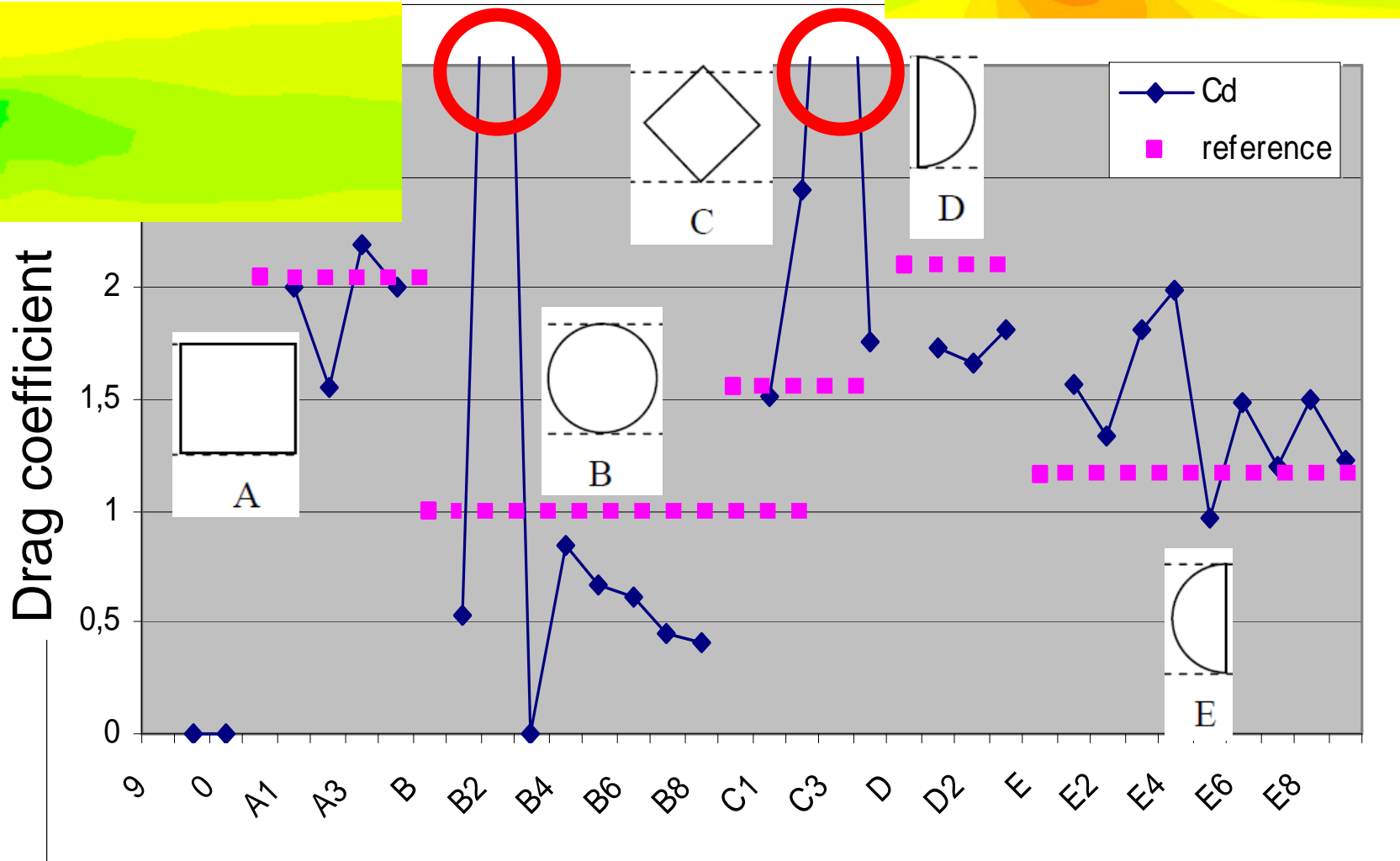
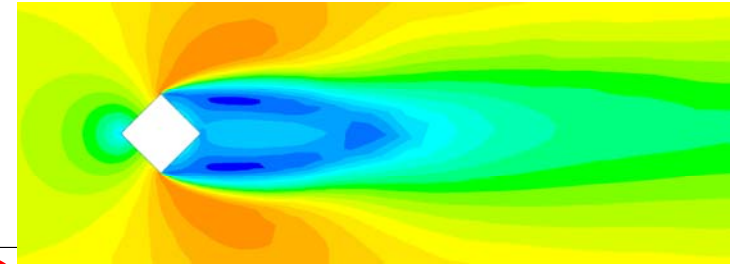
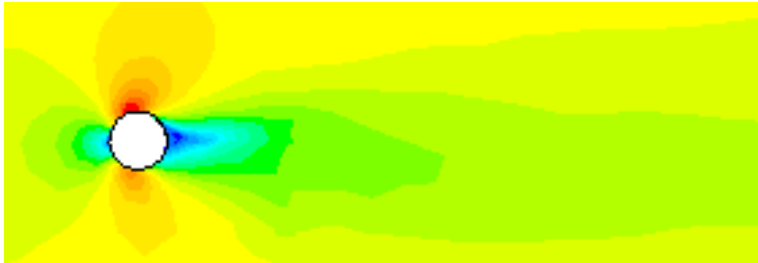
- Objective
 - Drag coefficient for a 2D object
- Before 16 April 12.00: Run the case – upload figures and drag



What can we learn?



Error?



What is the question?

- Drag coefficient

$$C_D = \frac{2D}{\rho U^2 H}$$

- $D=1.2, \rho=1.2, U=1, H=2 \rightarrow C_d=1.0$, not 2.0 or 1.2



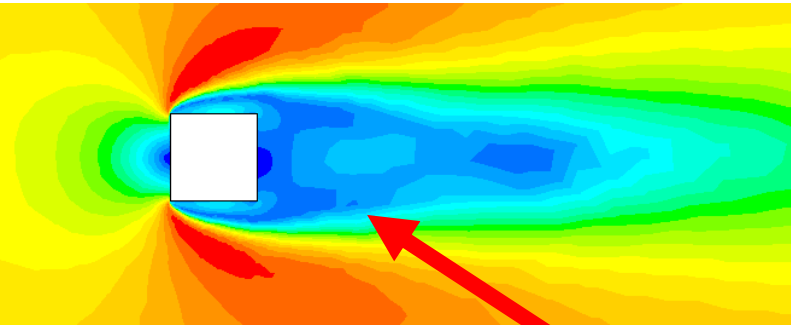
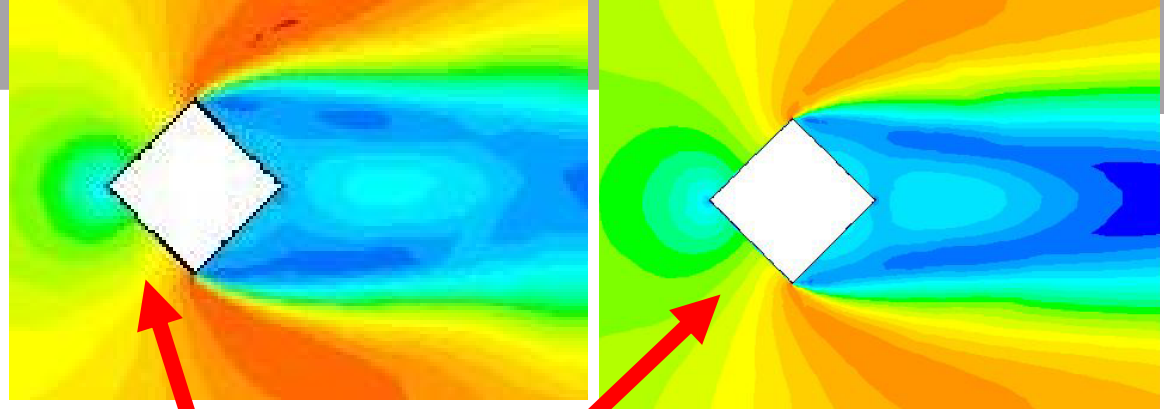
Forces

Zone object	Forces (n) Pressure (1.2247356 -0.056028187 0)	Viscous (-0.0087259216 4.0672516e-05)
Net	(1.2247356 -0.056028187 0)	(-0.0087259216 4.0672516e-05)

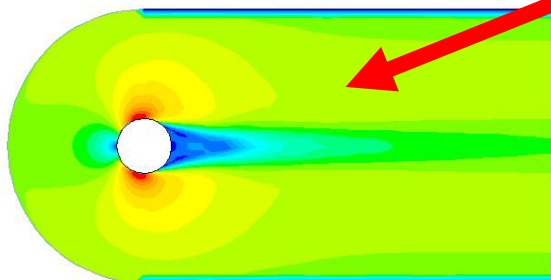
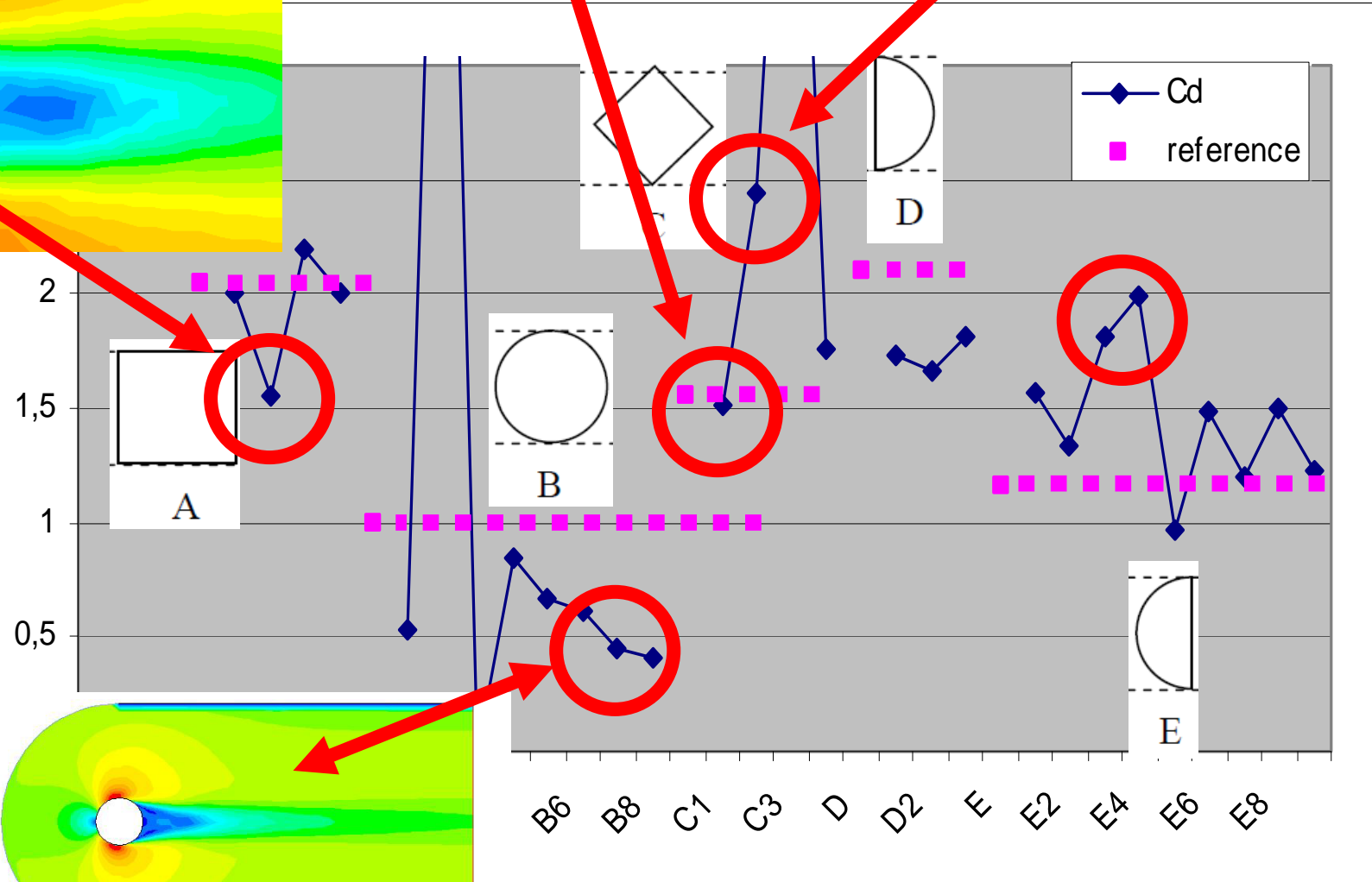
Forces - Direction Vector (1 0 0)

Zone object	Forces (n) Pressure 1.2247356	Viscous -0.0087259216	Total 1.2160097	Coefficients Pressure 1.9995684	Viscous -0.01424640
Net	1.2247356	-0.0087259216	1.2160097	1.9995684	-0.01424640

Quality?

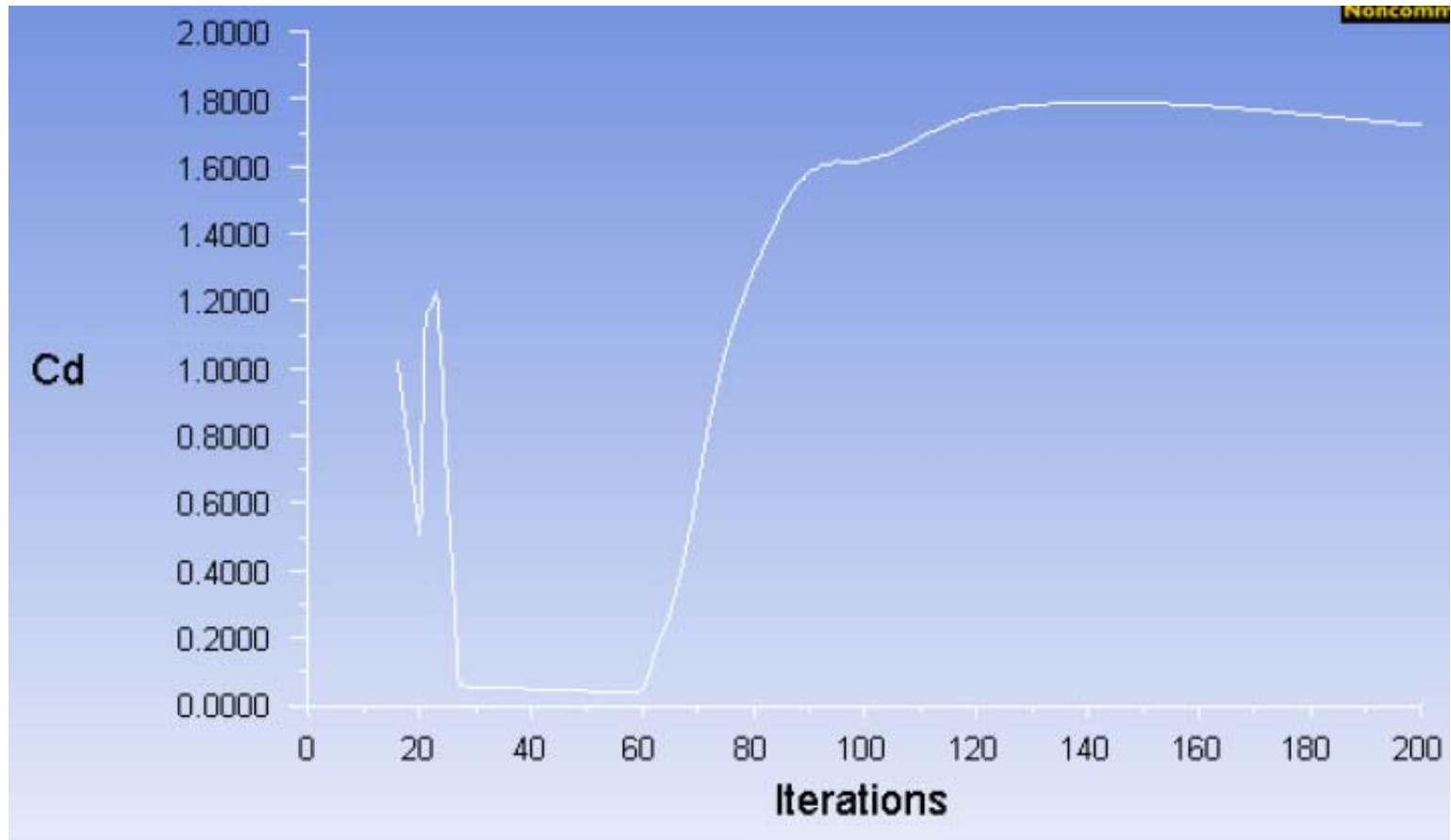


Drag coefficient

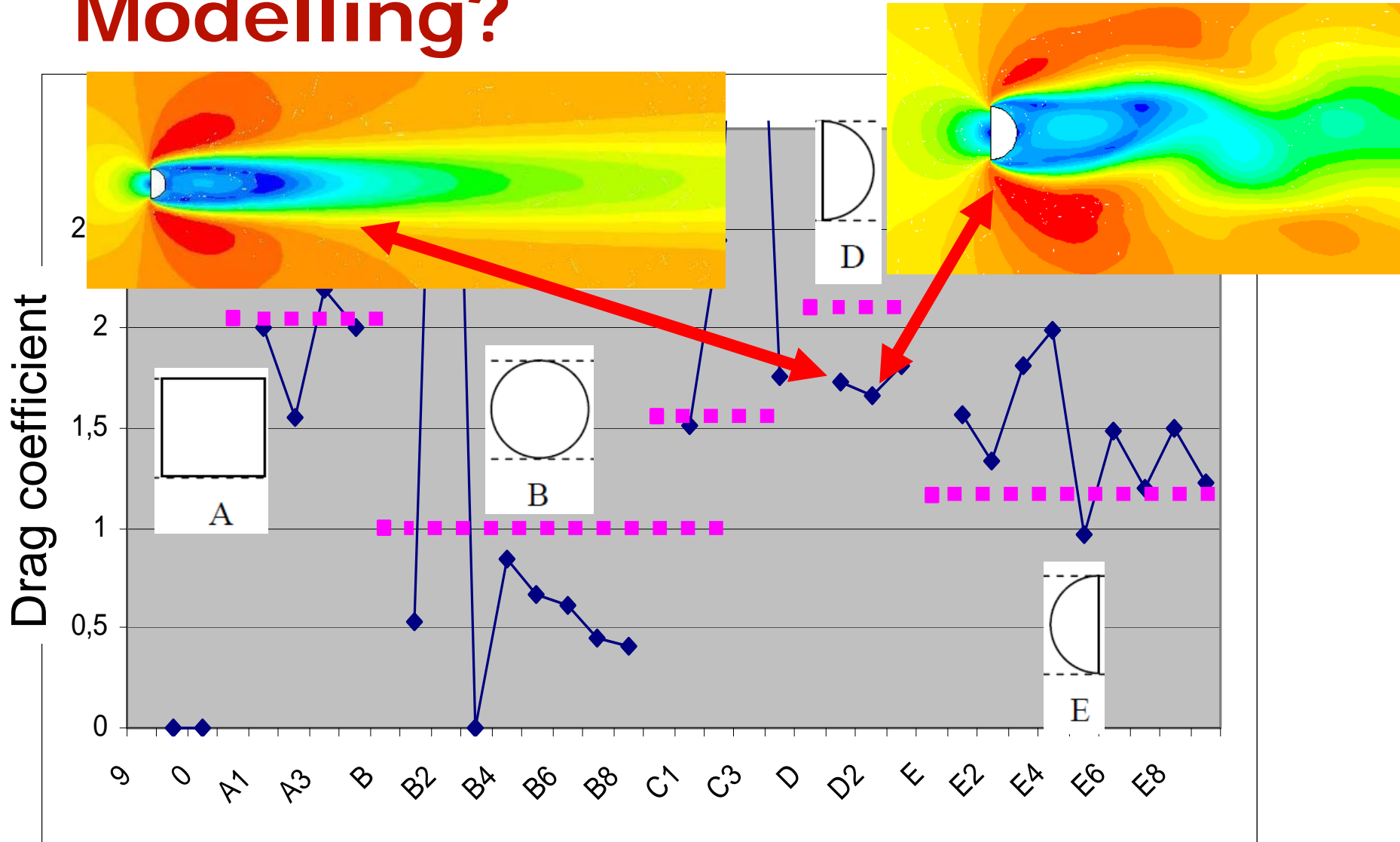


Convergence?

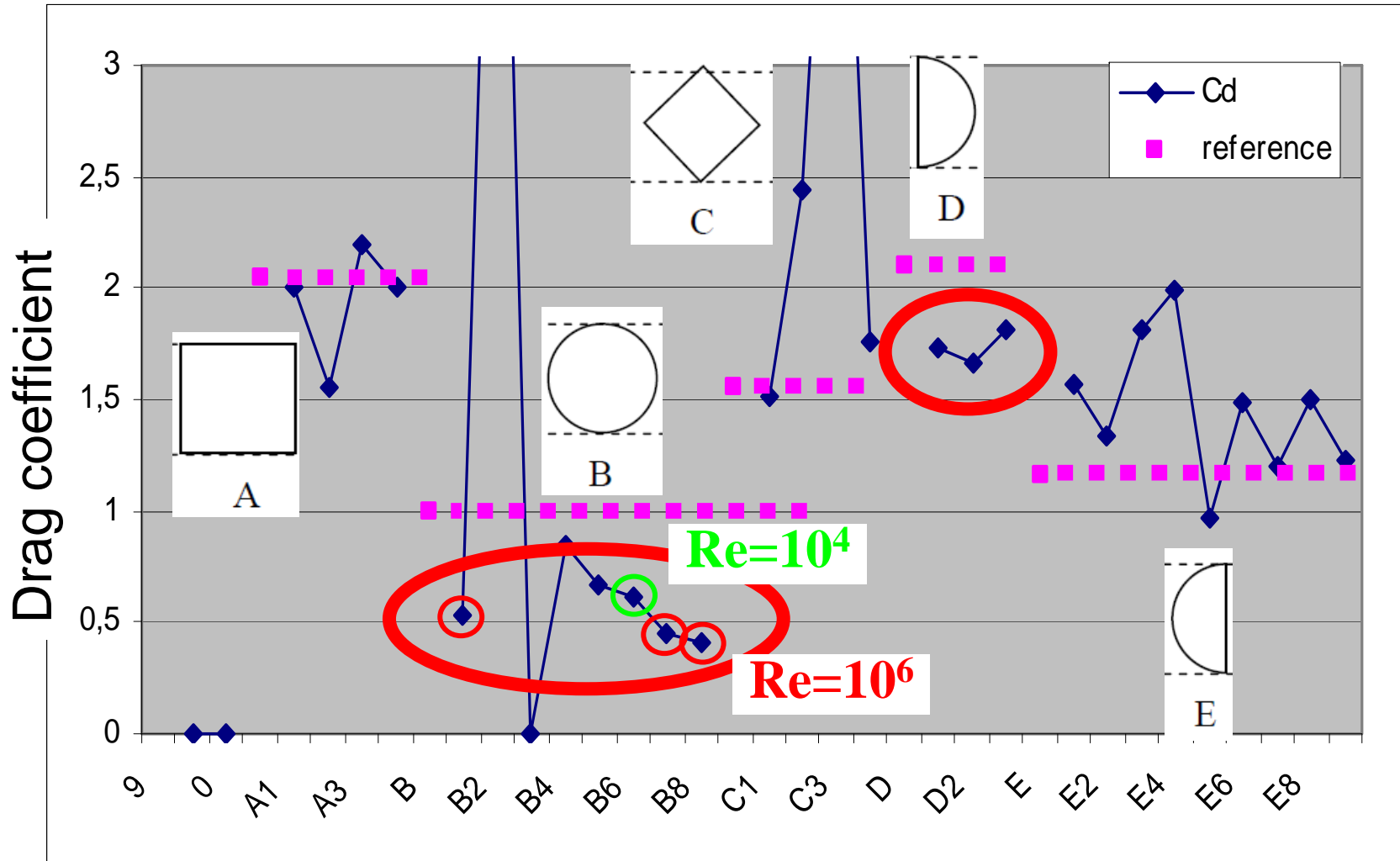
- Quantities of interest should not change with iteration



Modelling?



Systematic error?



Transition?

- Reynolds number $L=1\text{m}$, $U=1\text{m/s}$, air $\rightarrow Re \sim 10^5$



Laminar BL

