

# Turbulence on large computers

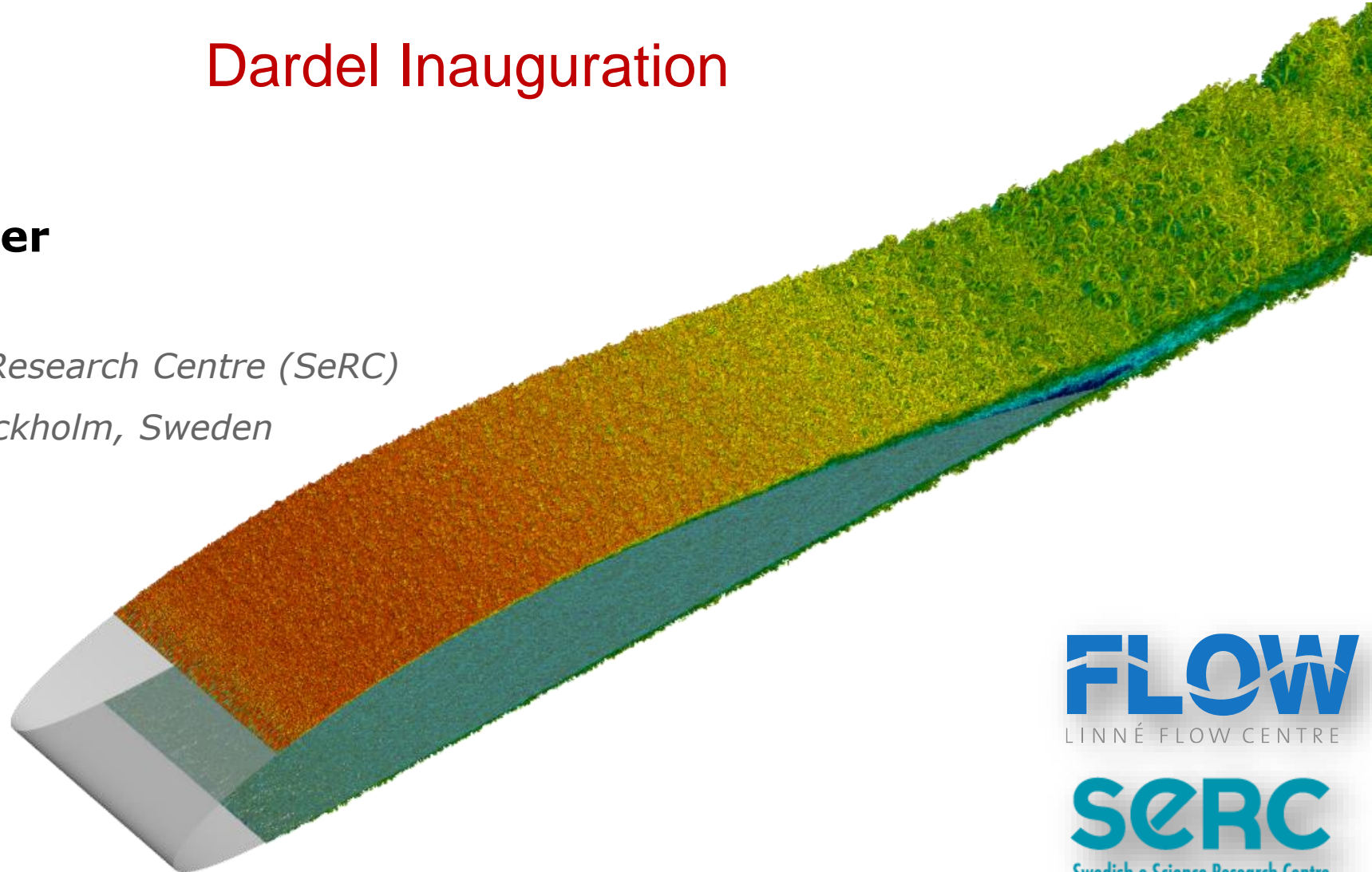
Dardel Inauguration

**Philipp Schlatter**

*FLOW Centre and*

*Swedish e-Science Research Centre (SeRC)*

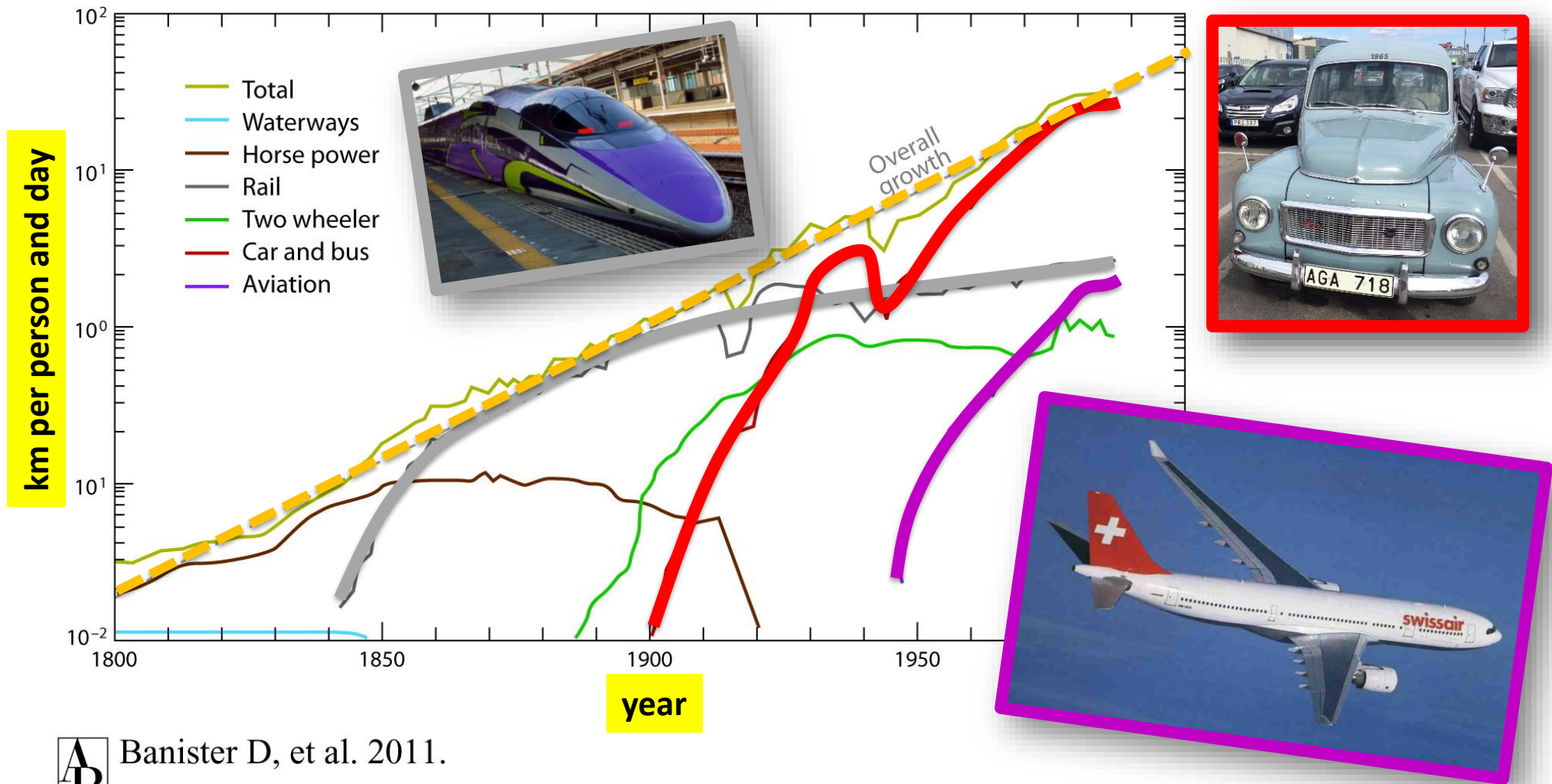
*KTH Mechanics, Stockholm, Sweden*



PDC, January 25, 2022

# Why turbulence?

Skin friction/drag reduction is the key for economically and ecologically more efficient transport



AR Banister D, et al. 2011.  
Annu Rev. Environ. Resour. 36:247-70




# Why are we here...?

- “When a sufficiently **advanced computer** becomes available, we believe it will **replace the wind tunnel** as the principal facility for providing aerodynamic flow simulations”
- “If past trends continue, such computer performance should **be available in the mid-1980s...**”

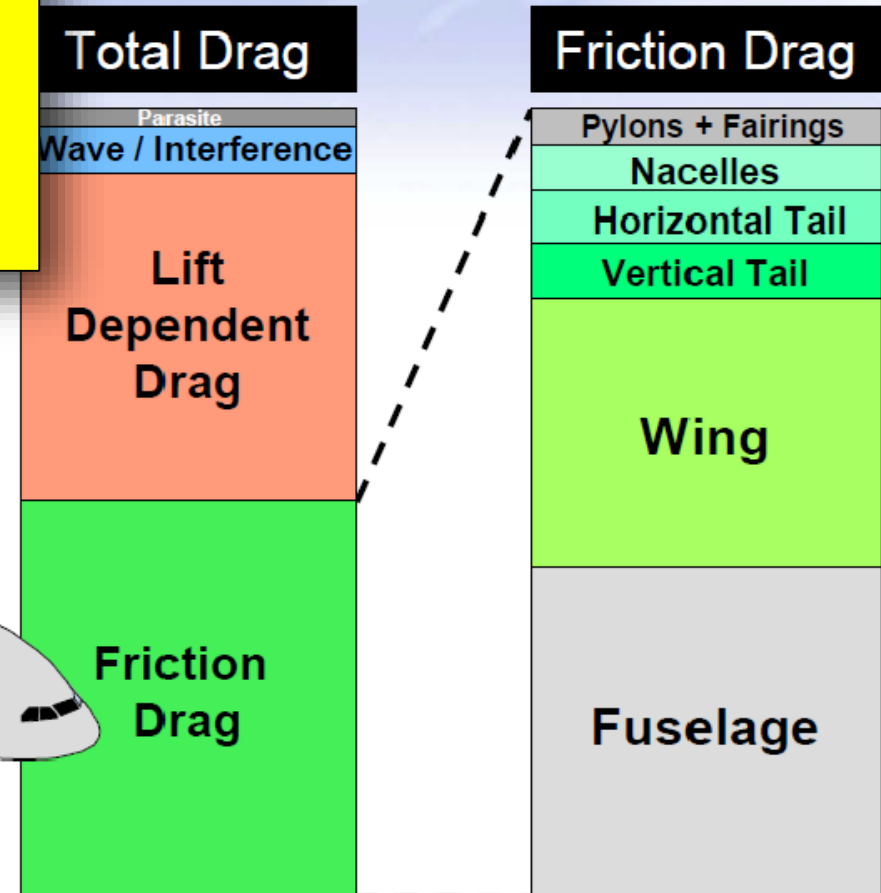
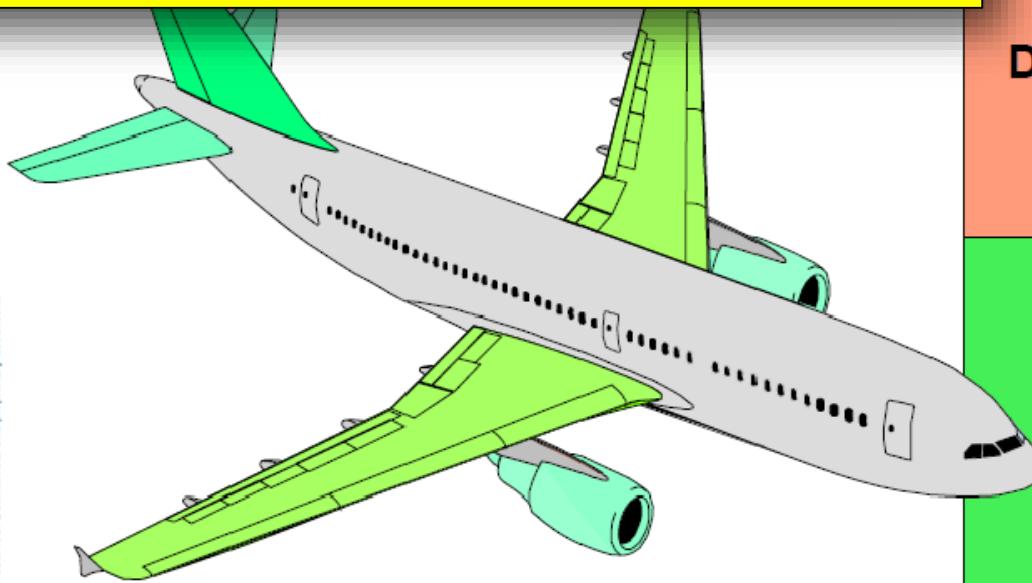
Chapman, D. R., Mark, H., Pirtle, M. W., “Computers vs. wind tunnels for aerodynamic flow simulations”, *Astronautics & Aeronautics* **13**(4):22-30, 1975 (NASA Ames)

# A Brief Diversion Into Aircraft Drag

A world of challenge & opportunity

 Typical break down of overall aircraft<sup>†</sup> drag by form & component

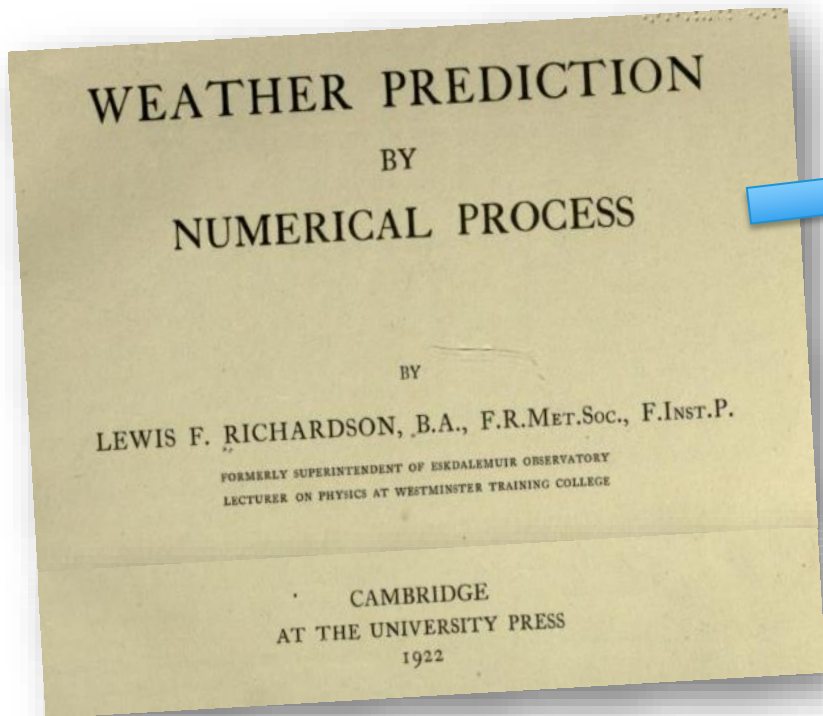
An Airbus 320 cruising at 250 m/s at 10000m  
**Tetalith ( $4 \cdot 10^{15}$  Flops): 200 years**  
**Result in one week:  $4 \cdot 10^{19}$  Flops (40 EFlops)**  
(based on John Kim's estimate, TSFP-9, 2015)



† = Based on a typical A320

# Humble beginnings 100 years ago...

- Lewis Fry Richardson (1881-1953)



structure of the clouds is often very complex." One gets a similar impression when making a drawing of a rising cumulus from a fixed point; the details change before the sketch can be completed. We realize thus that: big whirls have little whirls that feed on their velocity, and little whirls have lesser whirls and so on to viscosity—in the molecular sense.

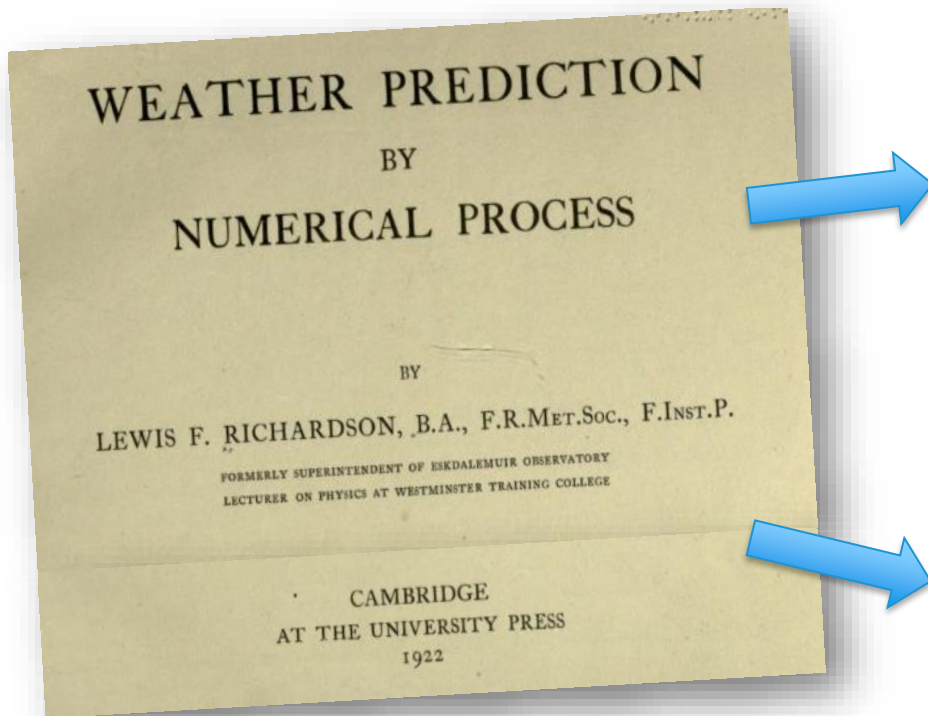
Thus, because it is not possible to separate eddies into clearly defined classes according to the source of their energy; and as there is no object, for present purposes,



play on Augustus de Morgan's famous paraphrasing<sup>3</sup> of Jonathan Swift

# Humble beginnings 100 years ago...

- Lewis Fry Richardson (1881-1953)



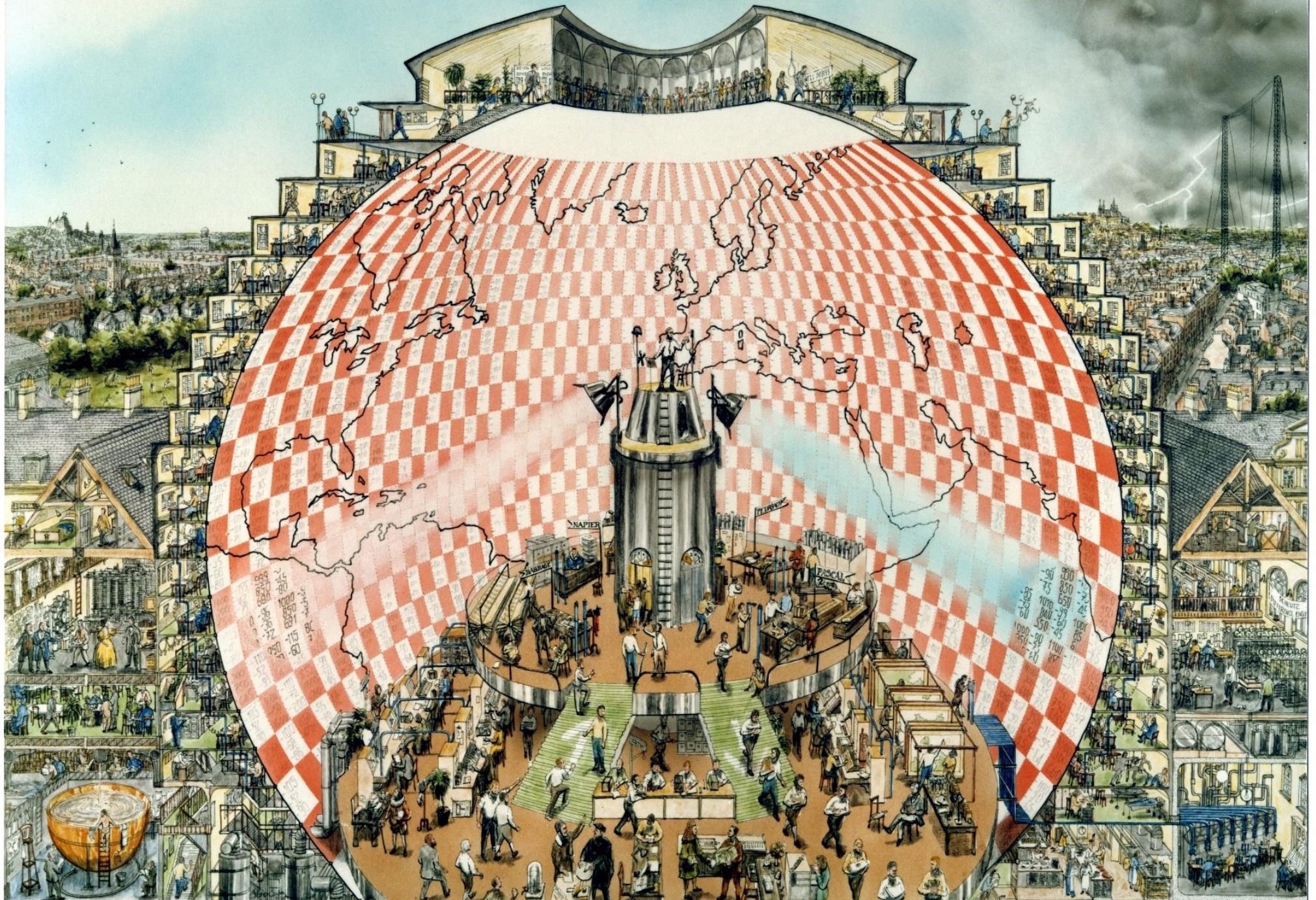
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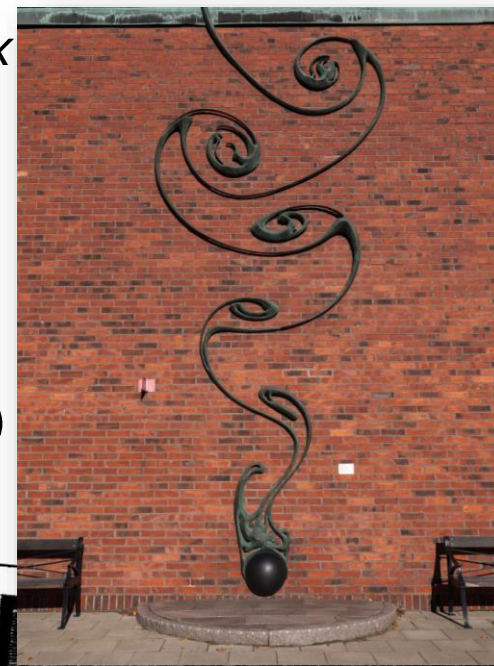
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"First simulations" 1920: Eight hours weather prediction in 6 weeks, using **2000 human computers**

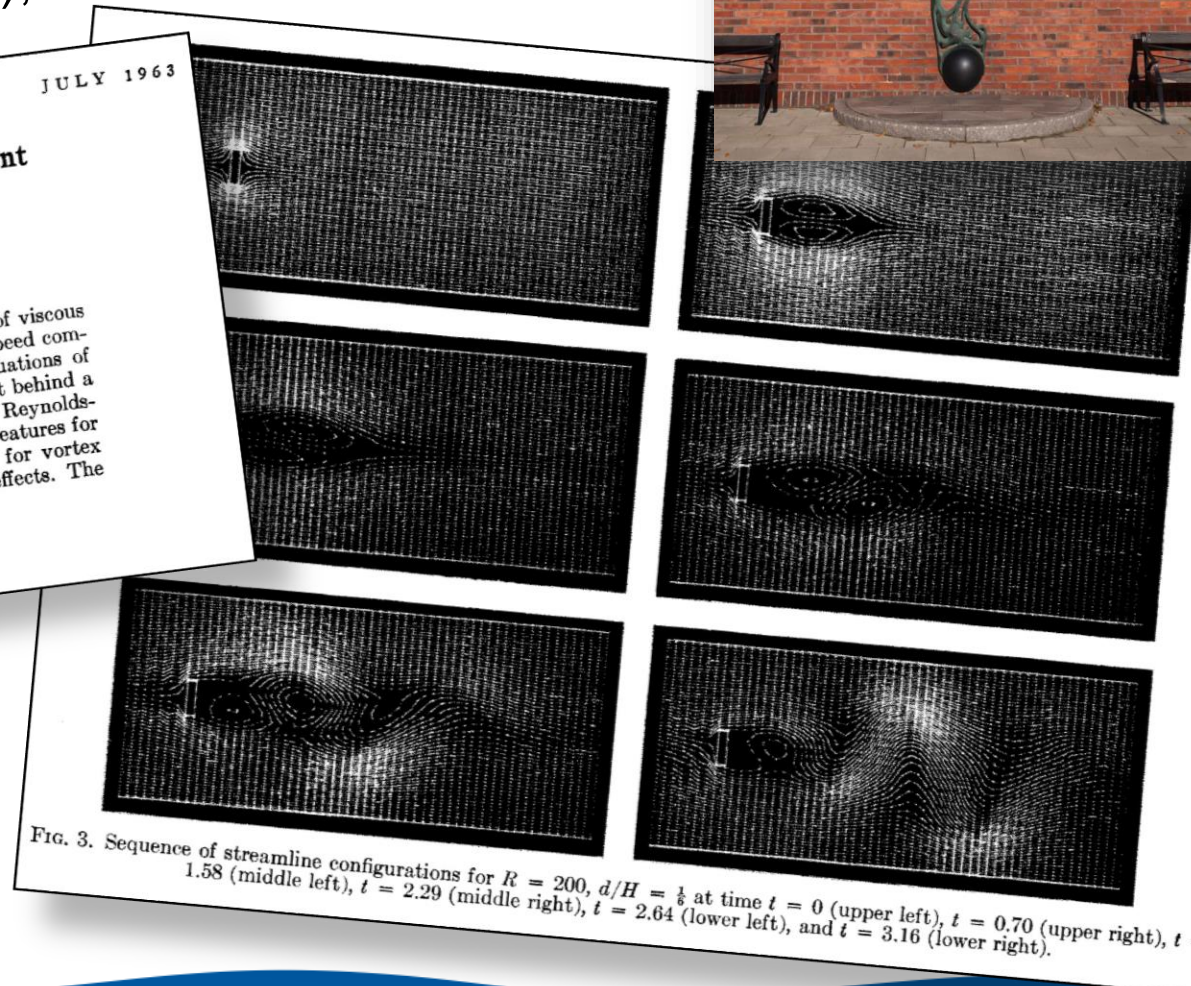
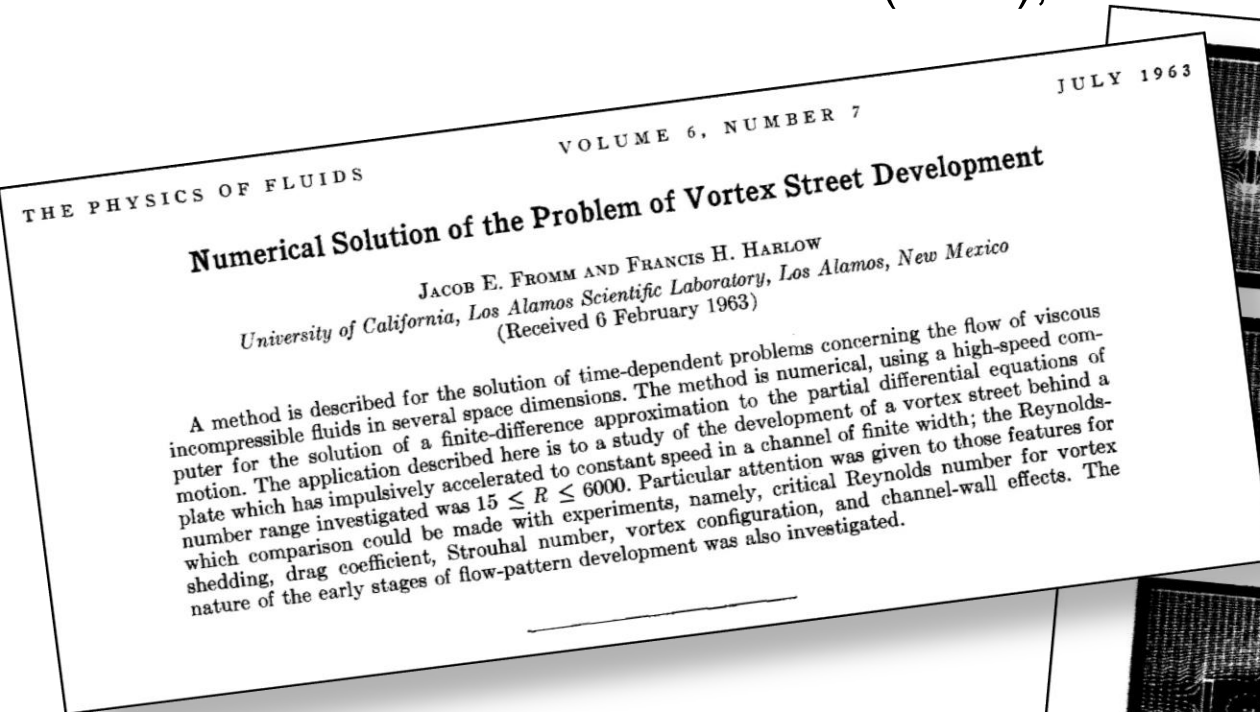
→ "Forecast-Factory"

The Weather Forecasting Factory by Stephen Conlin (1986)



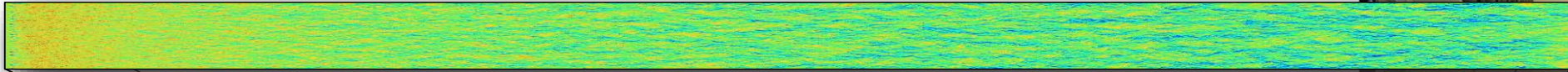
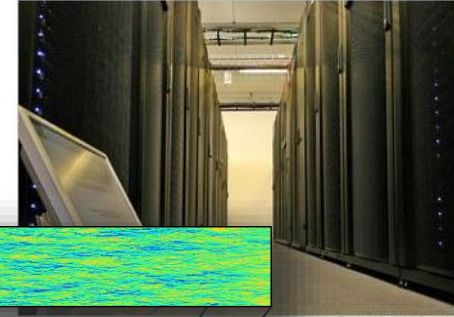


- Development of numerics (1940ies, 1950ies)
- Low- $Re$  cylinder wakes by Thom (1933), Kawaguti (1953) and Fromm & Harlow (1963), Los Alamos



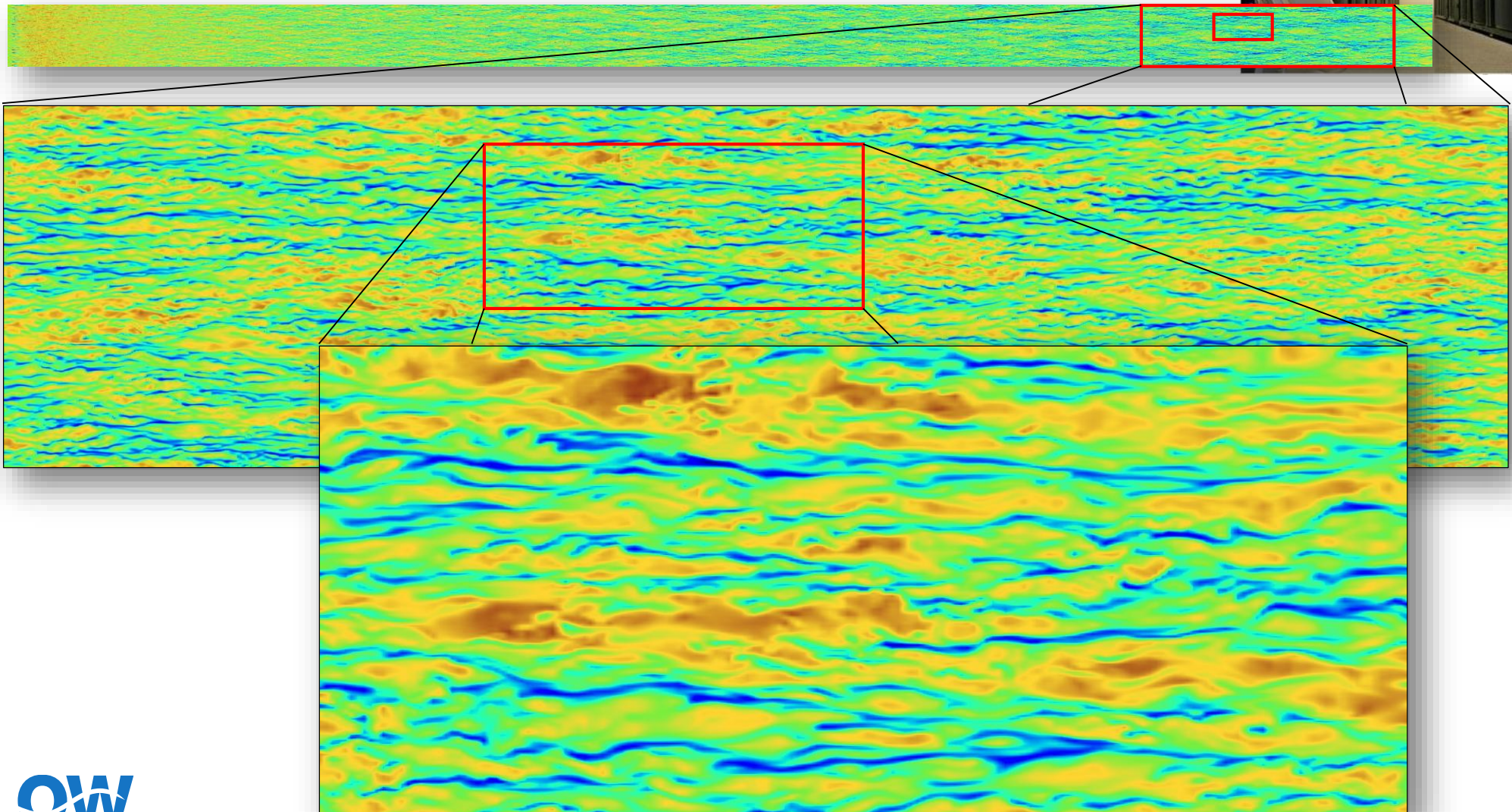
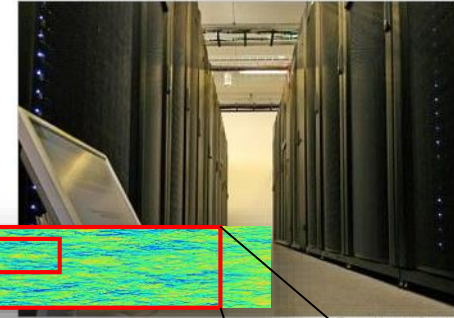


# Turbulent flow close to solid walls...



# Turbulent flow close to solid walls...

simulation result

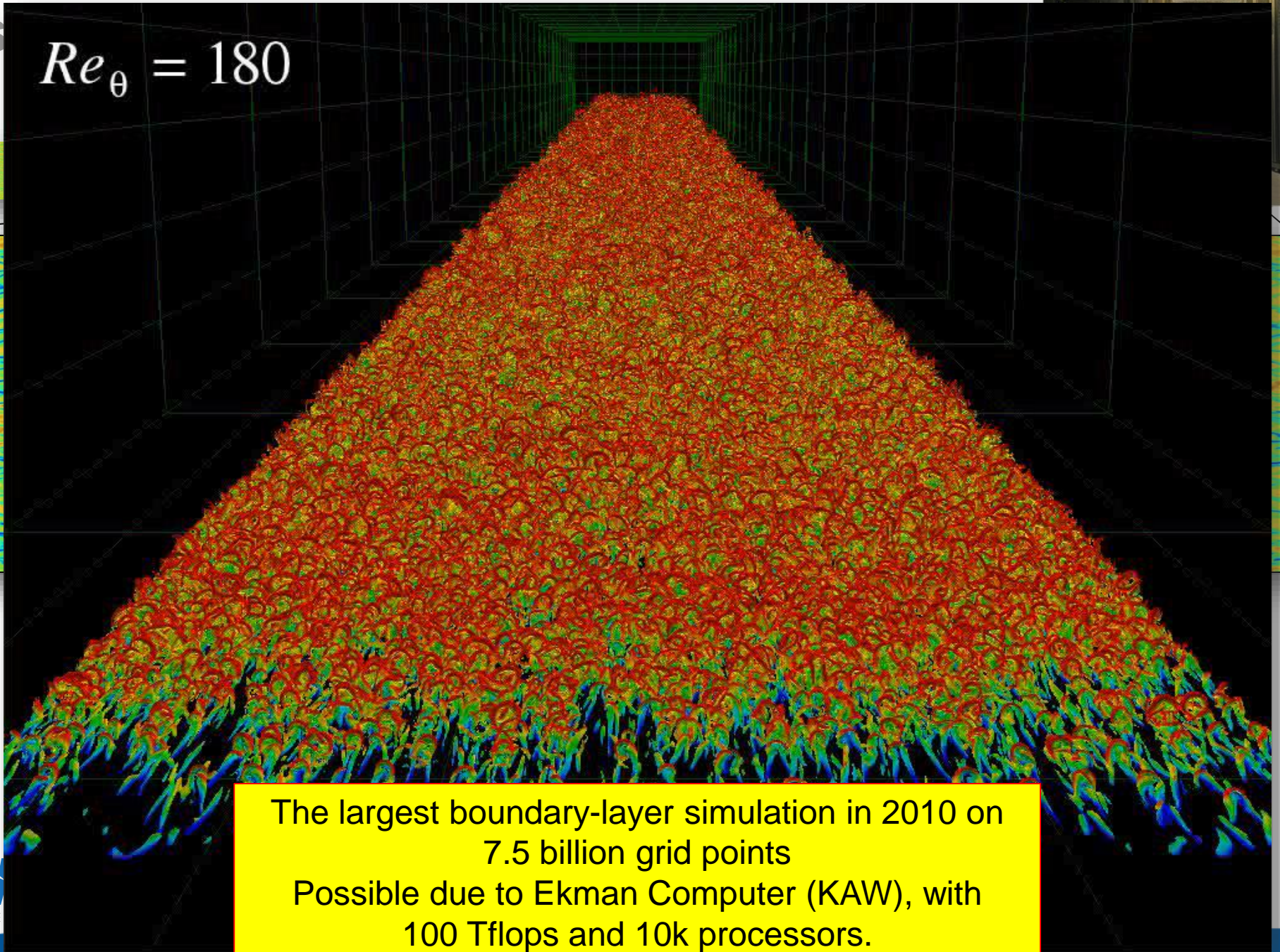
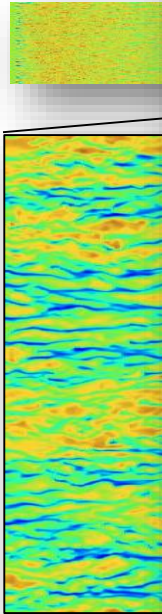




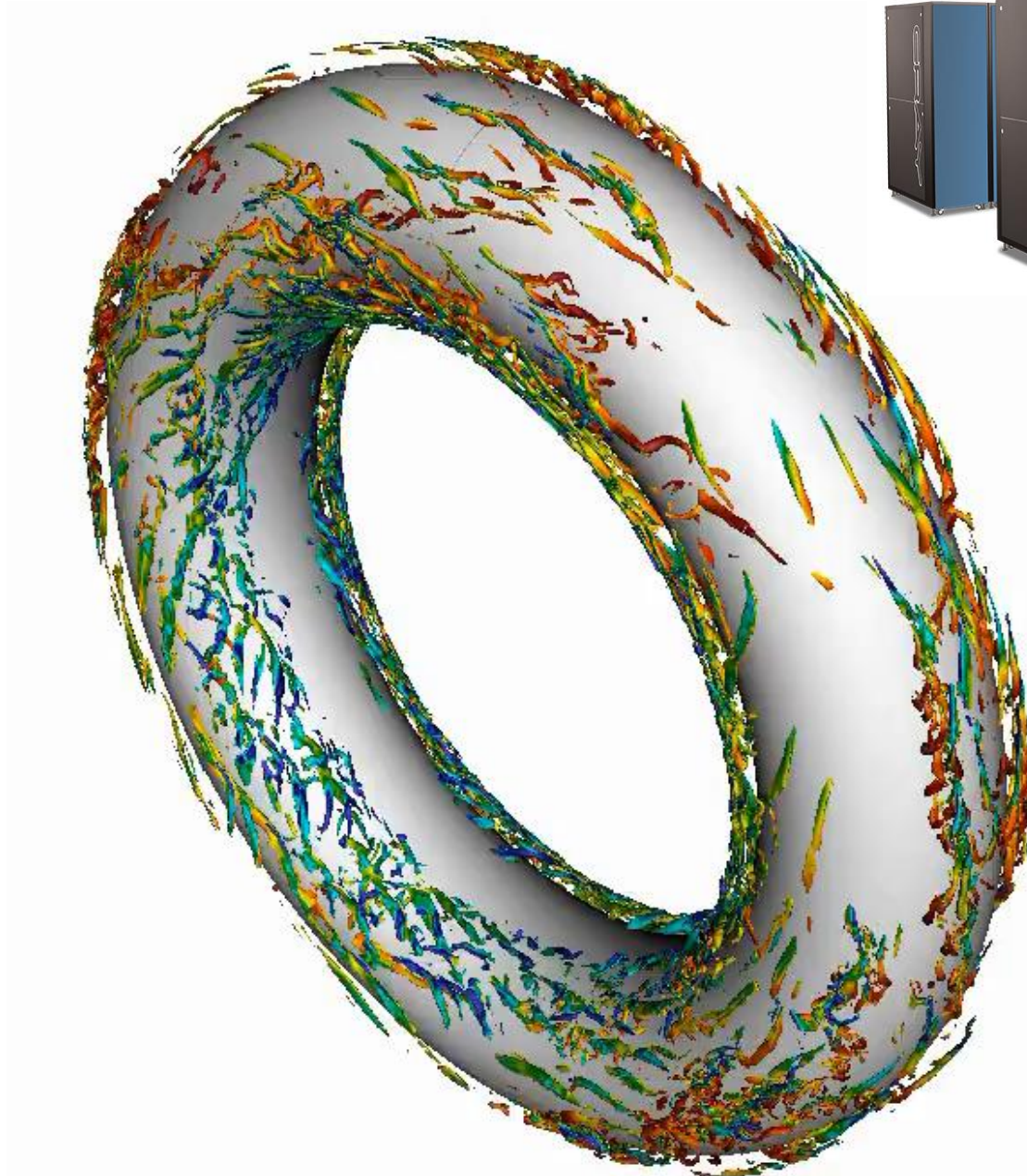
# Turbulent flow close to

S

$$Re_{\theta} = 180$$



The largest boundary-layer simulation in 2010 on 7.5 billion grid points  
Possible due to Ekman Computer (KAW), with 100 Tflops and 10k processors.



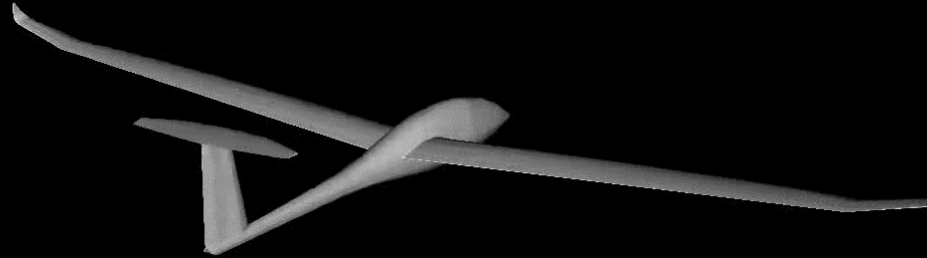
$$Re_{\tau} = 360$$

$$\kappa = 0.3$$

# DNS of flow around a NACA4412 wing section; $Re_c=400\ 000$ and $AoA=5^\circ$

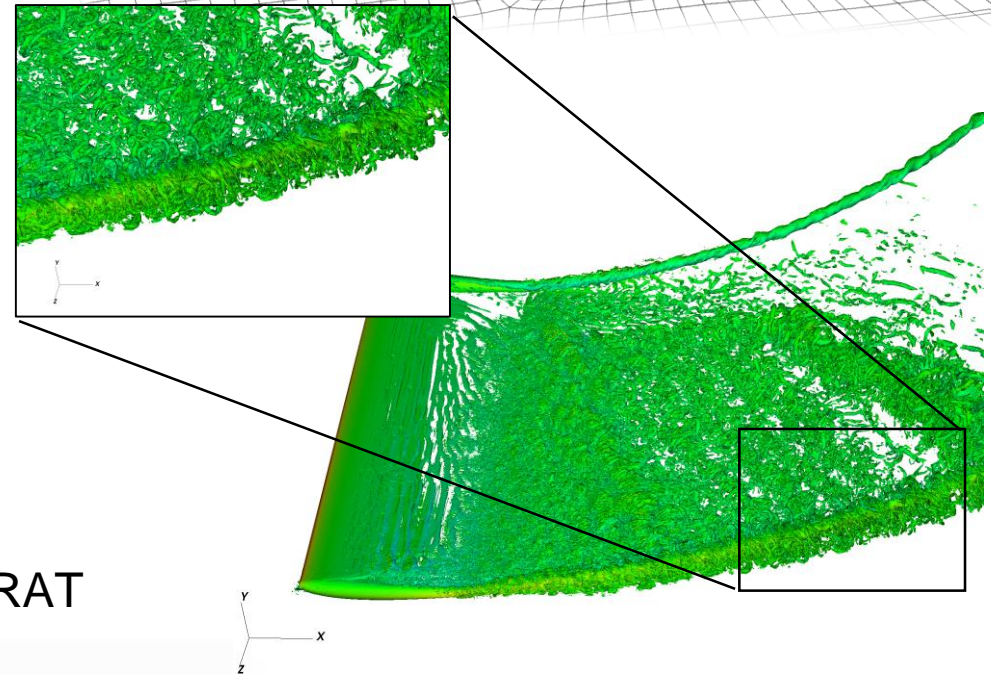
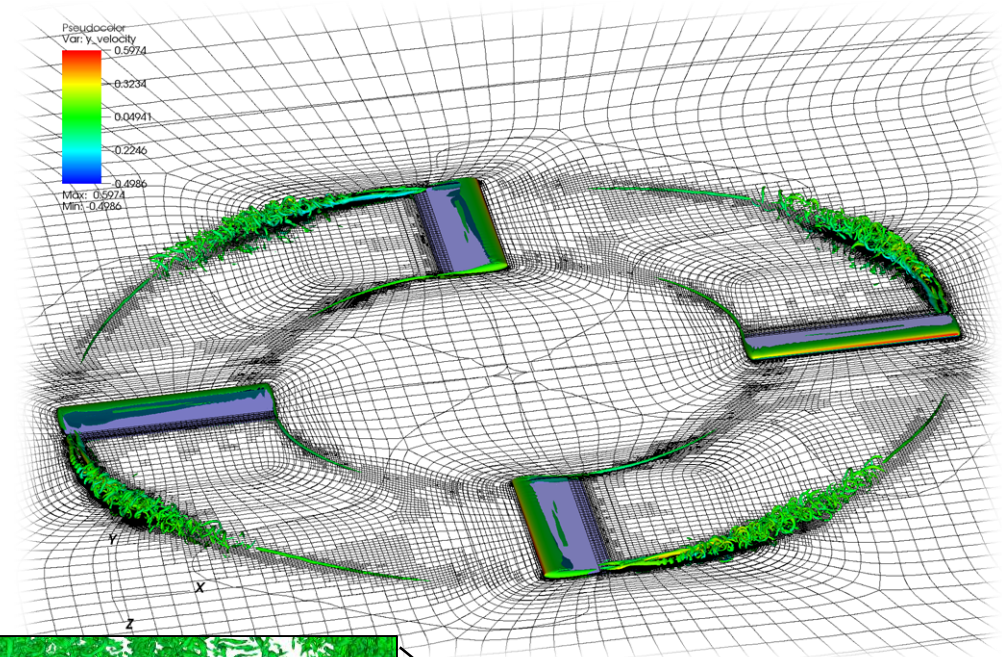


High-order methods are finding their way into aircraft design procedures, providing accuracy and reduced design risks, particularly for turbulent flows with regions of flow separation.



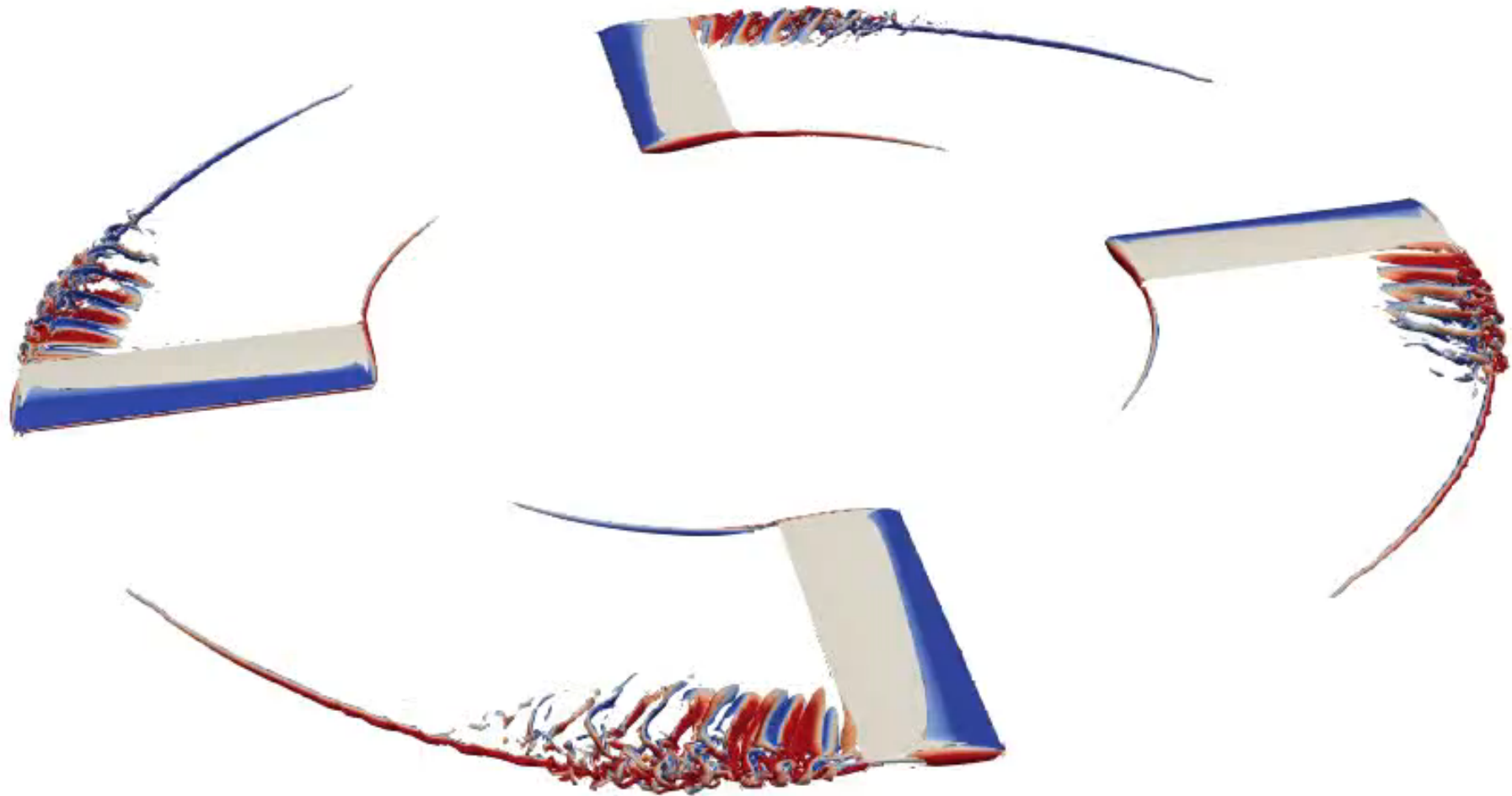
# DNS Rotating Propeller

- Physical details:
  - NACA0012, 5 degrees angle of attack, rounded wingtips
  - $Re=10k, 100k$  (based on chord and tip speed)
- Simulation details:
  - 1.1 million elements  
polynomial order 5
  - 300 million grid points
- One complete rotation: 250k core hours
  - 10 days on 1024 cores
- Collaboration with CINECA via EXCELLERAT
- Video by Antonio Memmolo



$Re=100k$

# DNS Rotating Propeller ( $Re=10k$ )





# Nek5000 – Spectral Elements

- SEM code by **Paul F. Fischer**, Argonne National Lab, USA  
Open source: `nek5000.mcs.anl.gov`
- 80 000 lines of **Fortran 77** (some C for I/O), MPI (no hybrid)
- **Gordon Bell Prize 1999** for algorithmic quality and performance
- **KISS** ("Keep it simple, stupid") – world's most powerful computers have very weak operating systems
- **EU Projects on algorithms** (CRESTA, ExaFLOW, Exellerat, Admire...): adaptive meshing, **GPUs**, compression...



ExaFLOW

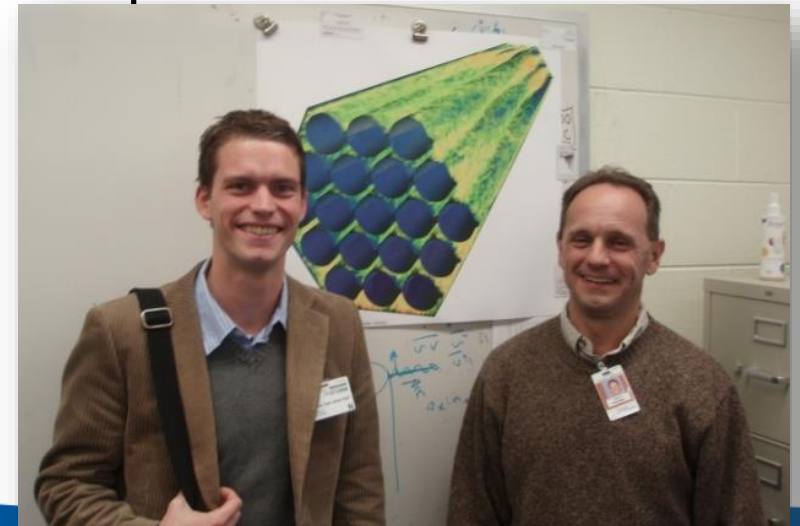


EXCELLERAT

ADMIRE

- Good scaling up to **>1,000,000** ranks

FLOW

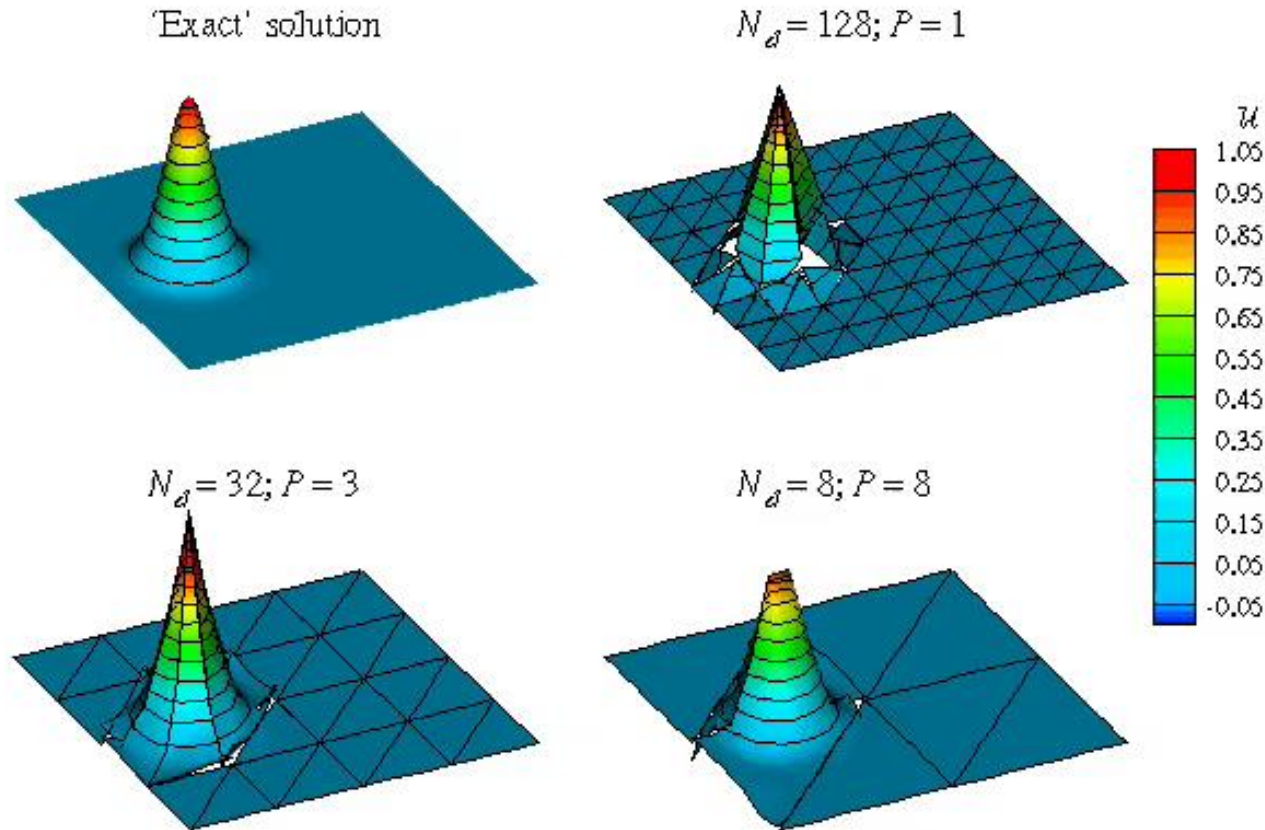




# Why Spectral Elements?

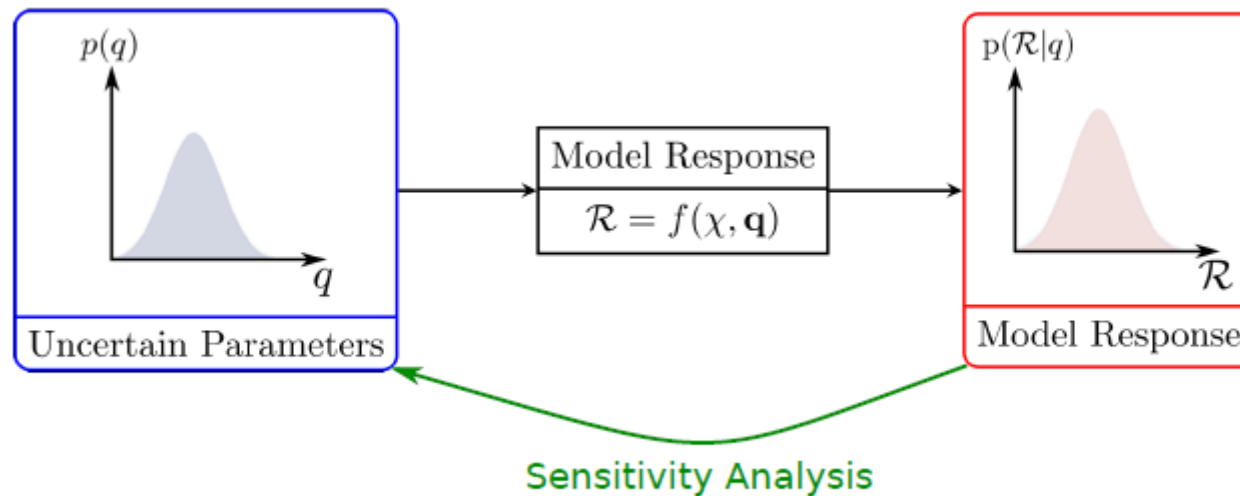
- Higher order  $p$  (vs. smaller grid spacing  $h$ ) means more work per core/communication: **"convecting cone"** [Gottlieb & Orszag 1977]

Time = 0



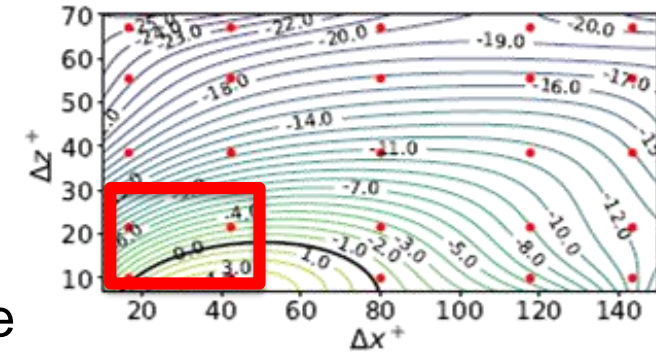
# Before we go any further; can we quantify accuracy?

- Let's look at standard channel flow,  $Re_\tau=300$ , DNS
- Two solvers: Nek5000 and OpenFOAM
- Consider varying  $\Delta x^+$ ,  $\Delta z^+$ , fixed  $\Delta y_w^+ = 0.5$  (fine)
- Quantities of interest (QoIs) for UQ problem:  $\langle u_\tau \rangle$ ,  $\langle u \rangle$ ,  $\langle u'_i u'_j \rangle$

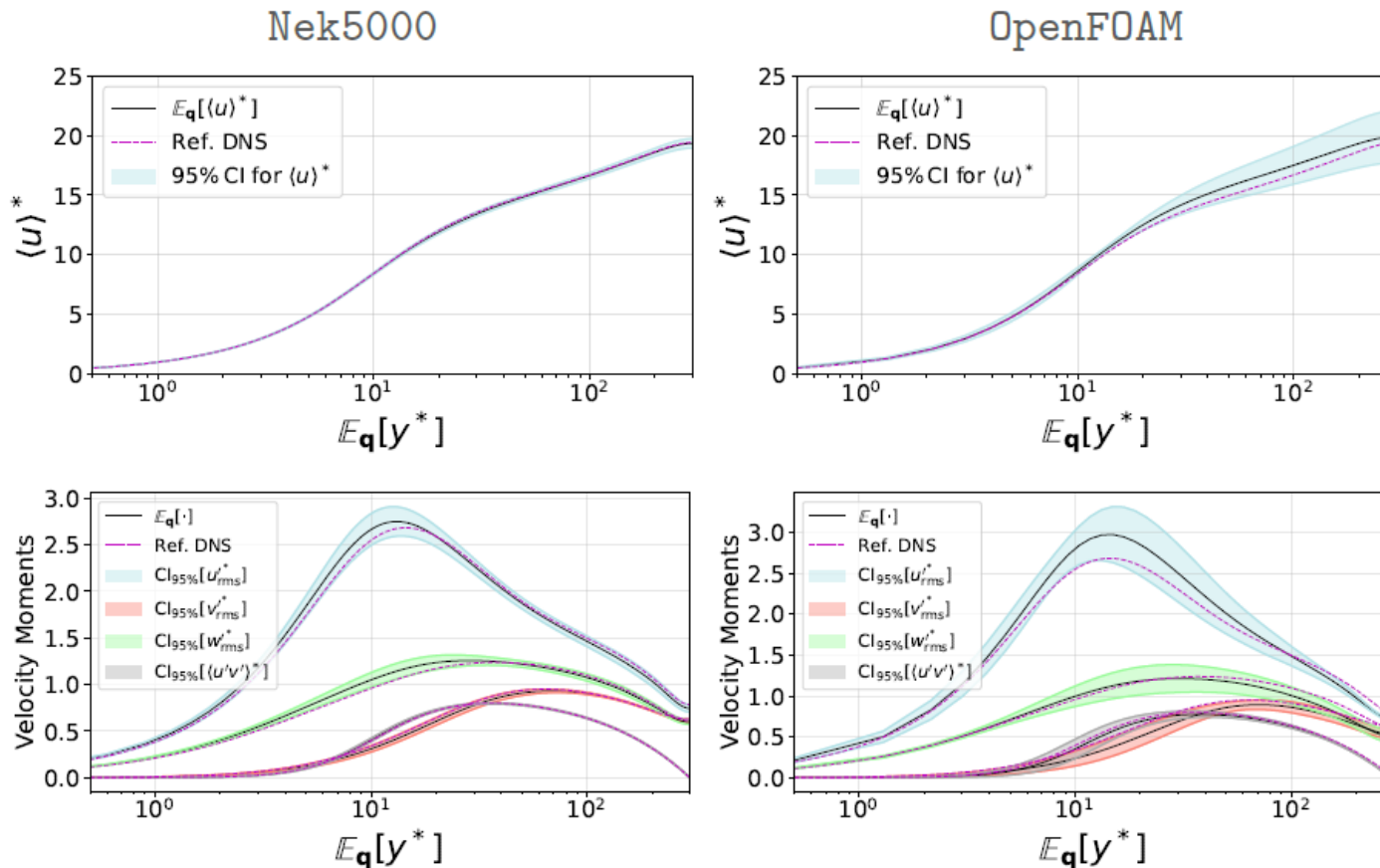


Ref. Smith 2013, Ghanem *et al.* 2017, Rezaeiravesh *et al.* 2021

# UQ: Robustness Channel Flow

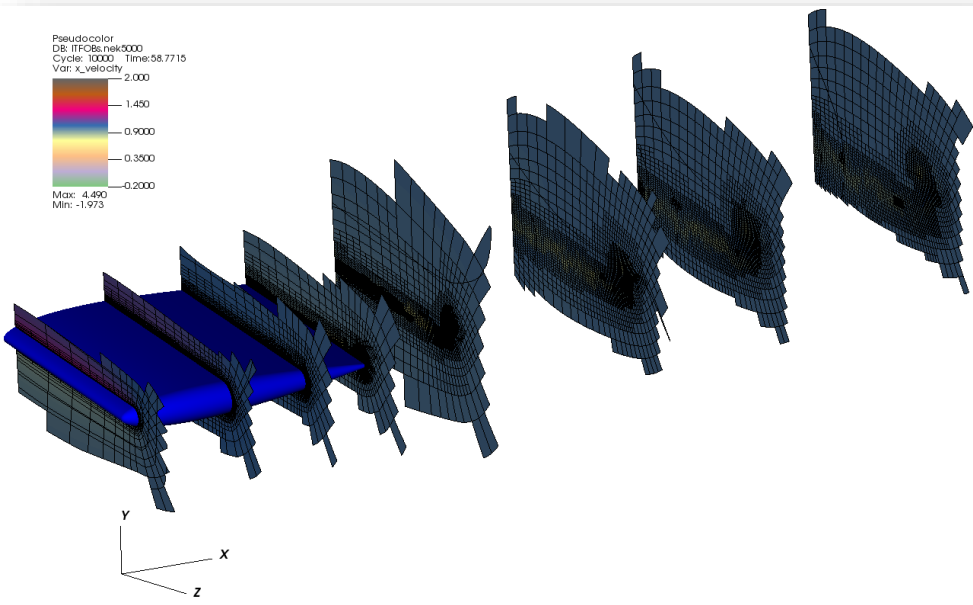


- How robust is a QoI with respect to variations of the numerical parameters? [Santner *et al.* 2003, Smith 2013]
- Uncertainty propagation by polynomial chaos expansion [Xiu&Karniadakis 2002]  $\Delta x^+ \sim \mathcal{U}[10, 50]$  and  $\Delta z^+ \sim \mathcal{U}[7, 30]$

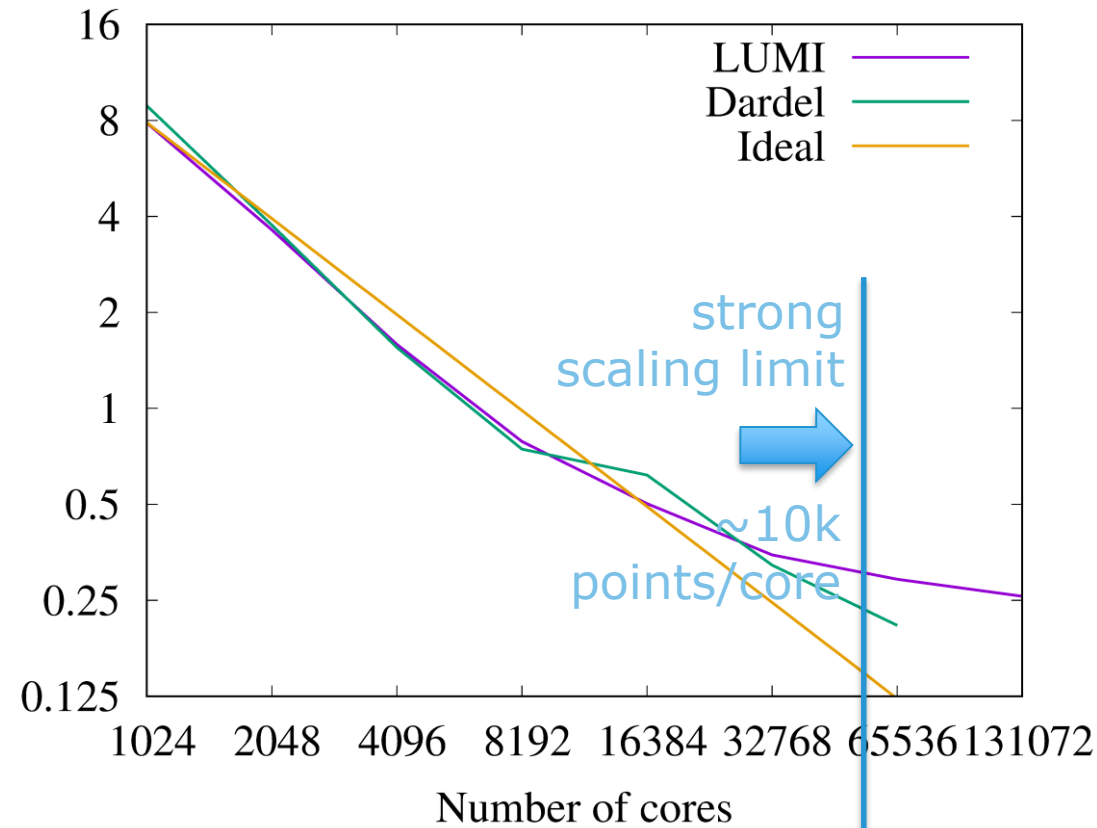


# Some scaling data Dardel pilot phase

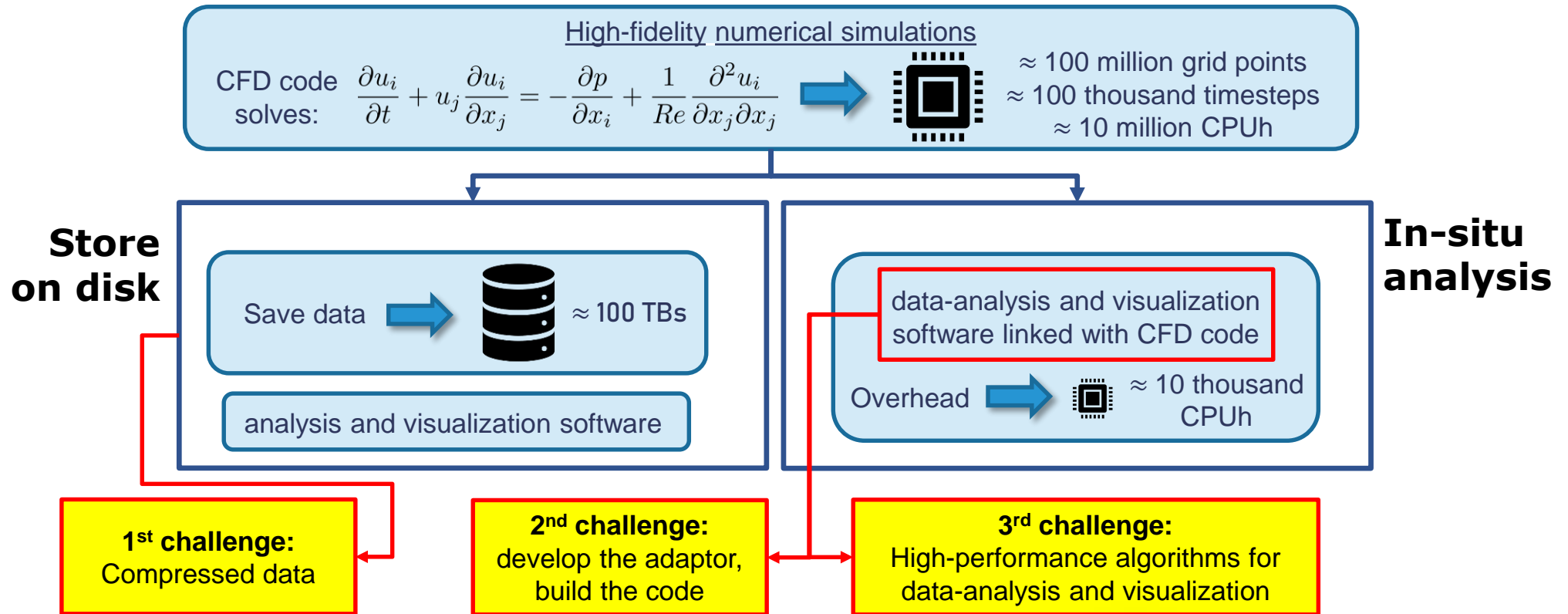
- Three-dimensional wing tip case, non-conformal meshes, 600k elements (500M grid points)



Time step duration (s)



# Main challenge: DATA



Multiple frameworks exist:

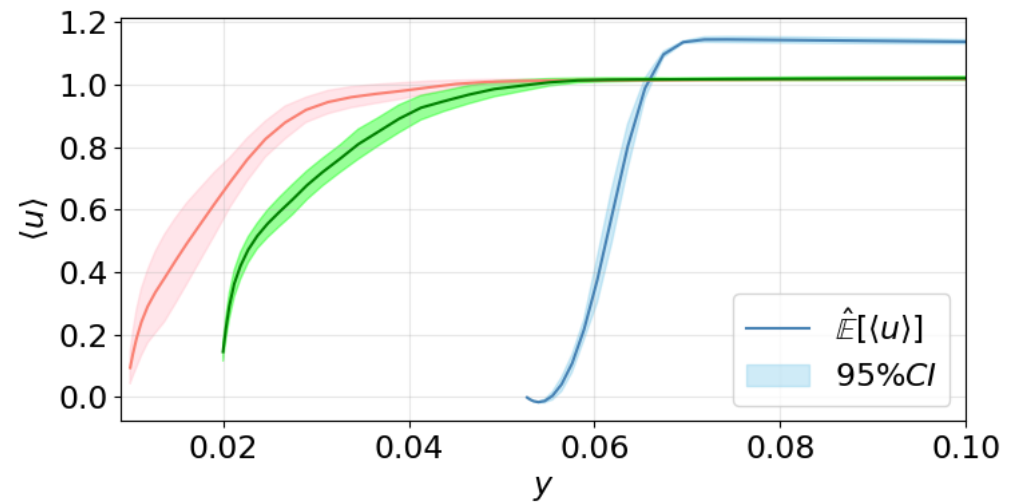
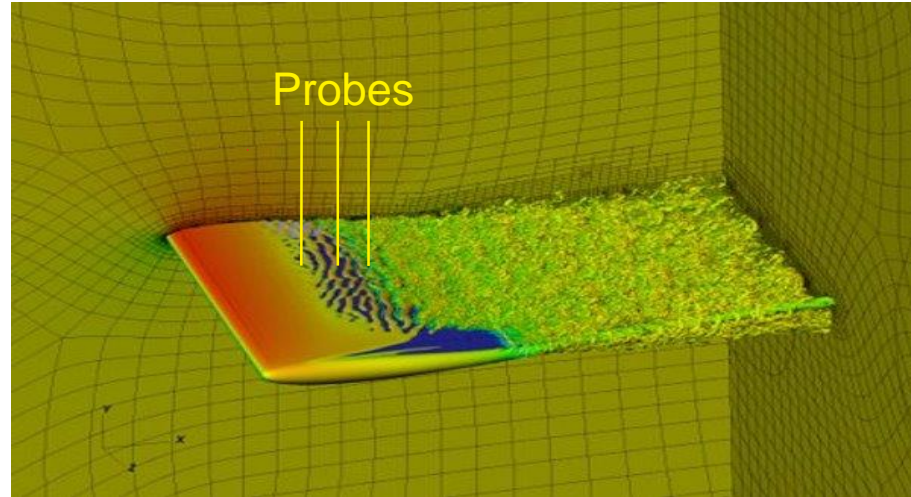
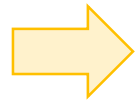
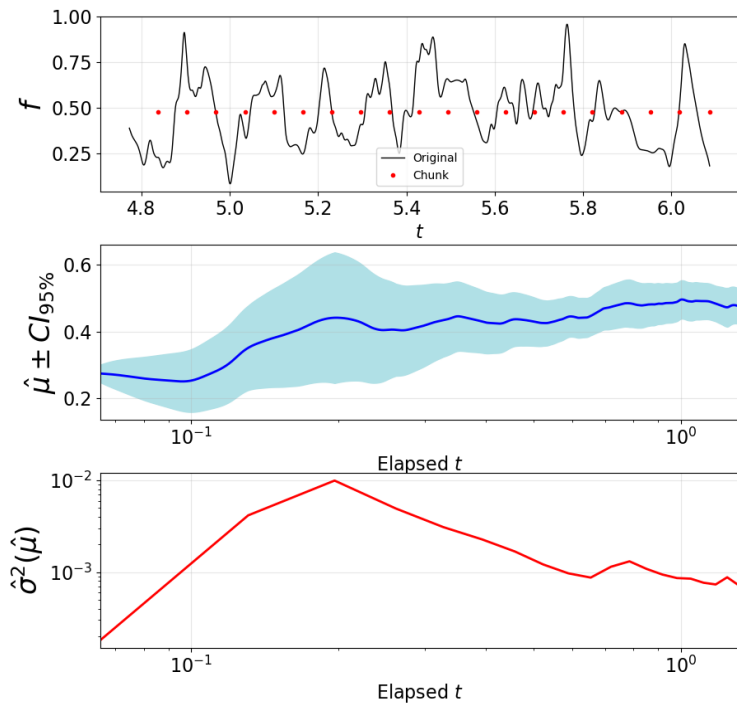
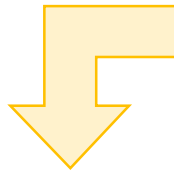
**Paraview / Catalyst**  
**ADIOS2**  
**VisIt / Libsim**  
**SENSEI**



Reliable Estimation of the uncertainty due to the finite time-averaging

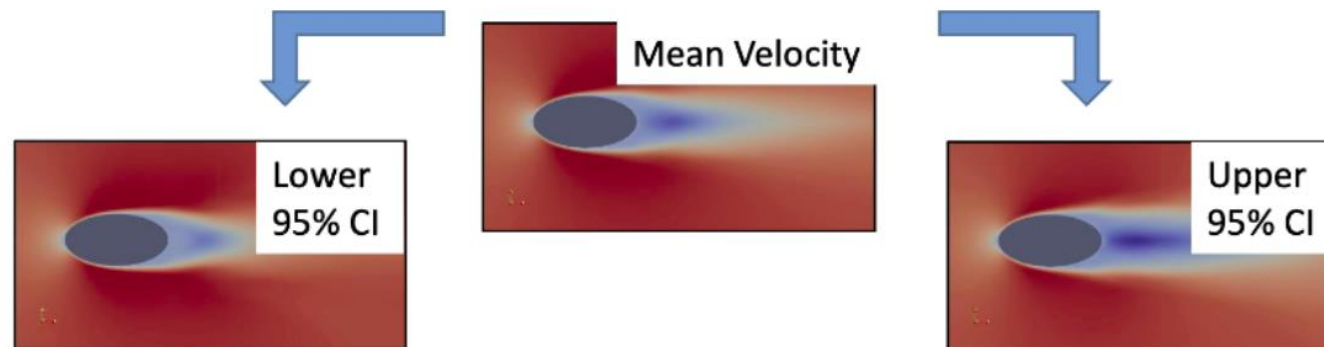
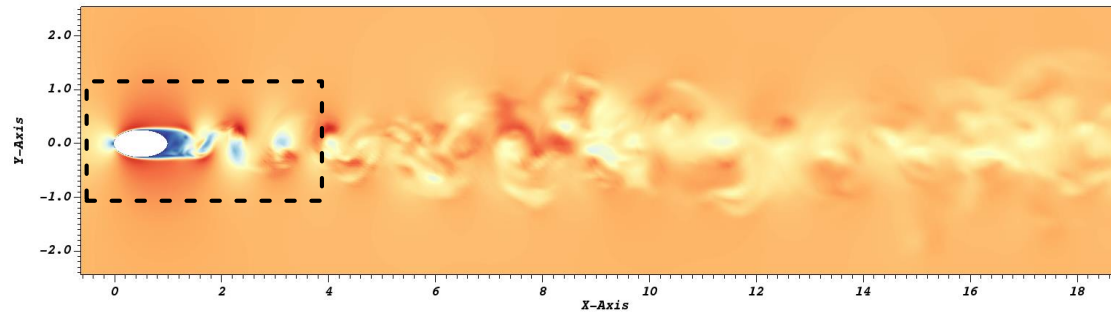
- Autoregressive model
- Modeled Autocorrelation Function

$$\hat{\mu} \sim \mathcal{N}(\mu, \sigma^2(\mu))$$



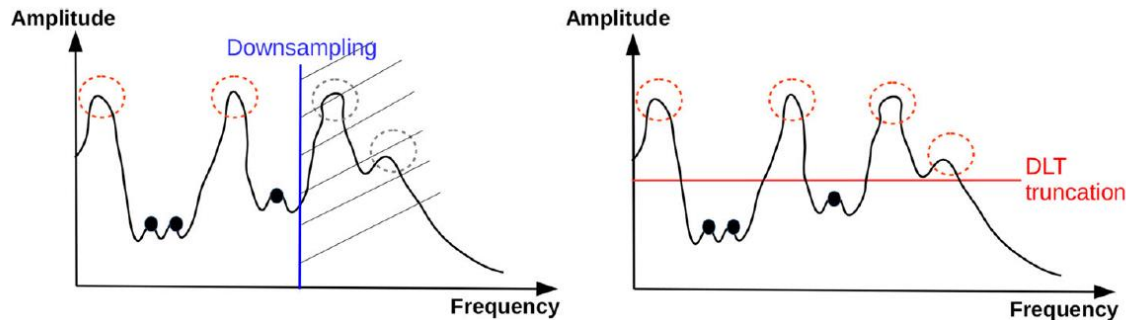
Mean velocity profiles and associated uncertainty

- **Updating algorithms** for sample-mean estimator, batch means, variance, and autocorrelations
- No need to dump the time-series samples!
- Collaboration within EXCELLERAT (KTH and Fraunhofer-SCAI)

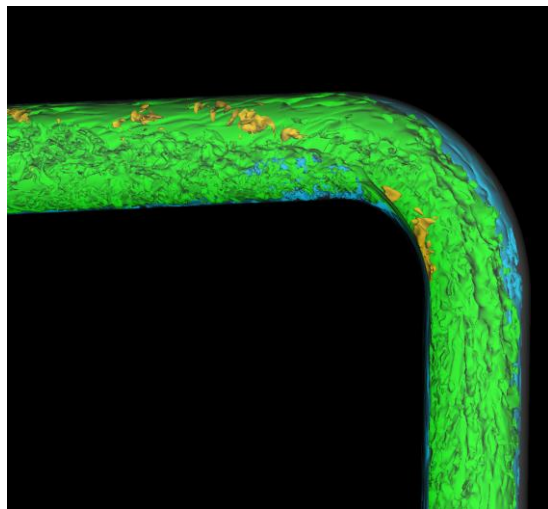


# Lossy compression of turbulence fields

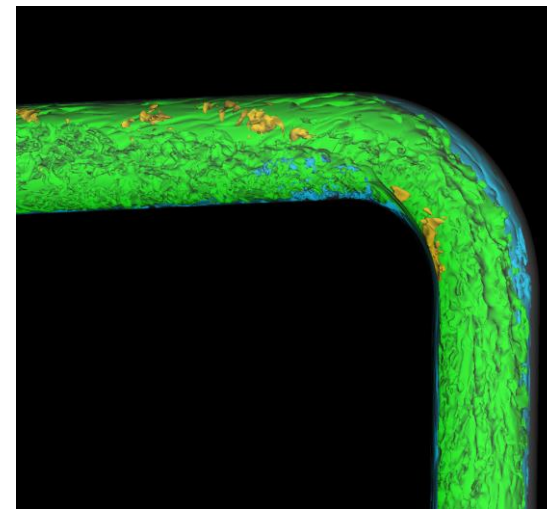
- A lot of entropy in turbulence, loss-less compression does not help much
- JPG-inspired algorithm in spectral space (Otero *et al.* 2018)



- Runtime encoding using ADIOS2 library in collaboration with Max Planck Computing and Data Facility



In-situ  
compression



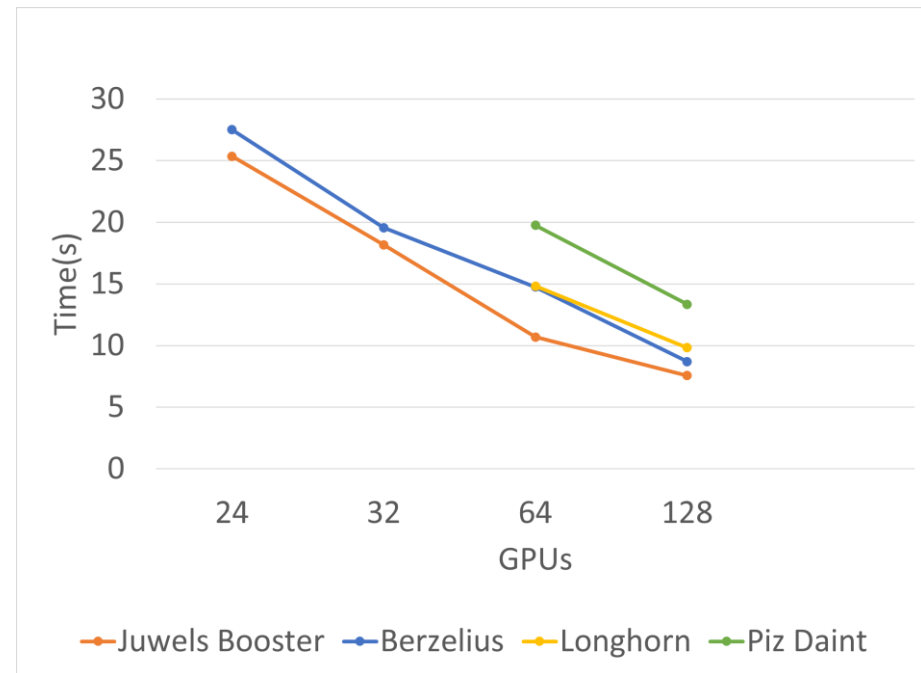
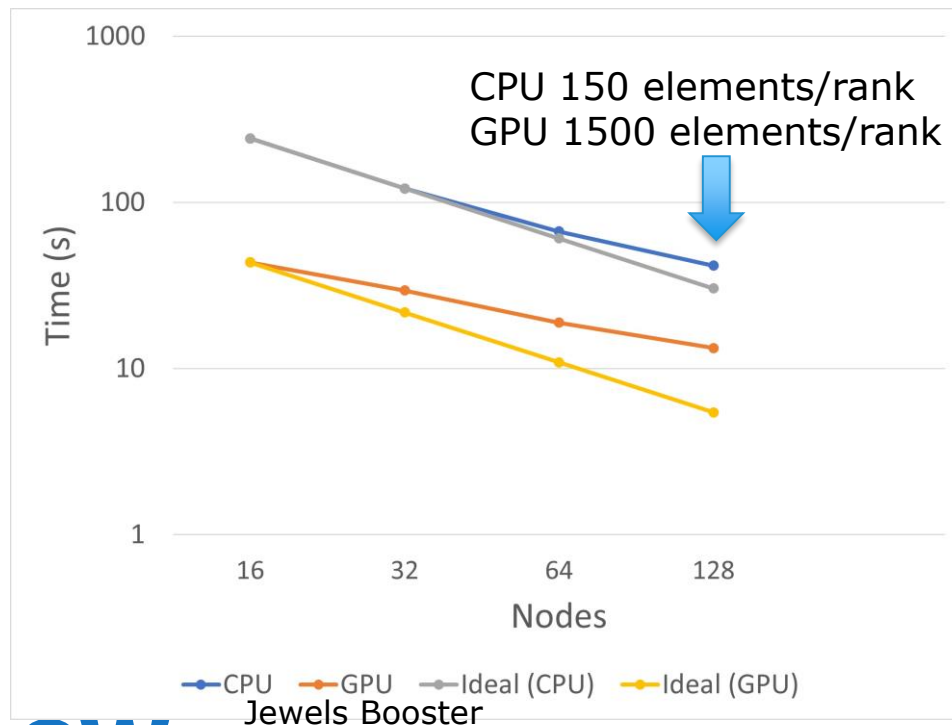
11% of data,  
error 0.01%



# What about Dardel phase 2?

## ■ Nek5000 results – CPU vs GPU scaling

- $Re_\tau = 550$ , max polynomial order 9
- $Re_\tau = 360$ , max polynomial order 9



# Neko: Portable Spectral Element Framework

- High-order spectral element flow solver
  - Matrix-free formulation, small tensor products
  - Modern object-oriented approach
    - > Based on the abstract problem,  $a(u, v) = L(v) \quad \forall v \in V$
    - > Modern Fortran, F2008/F2018 (coarrays!)

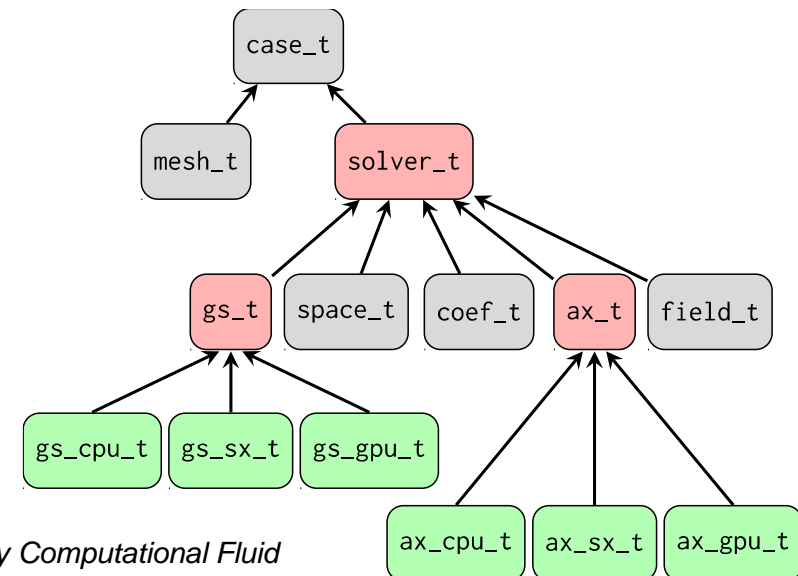
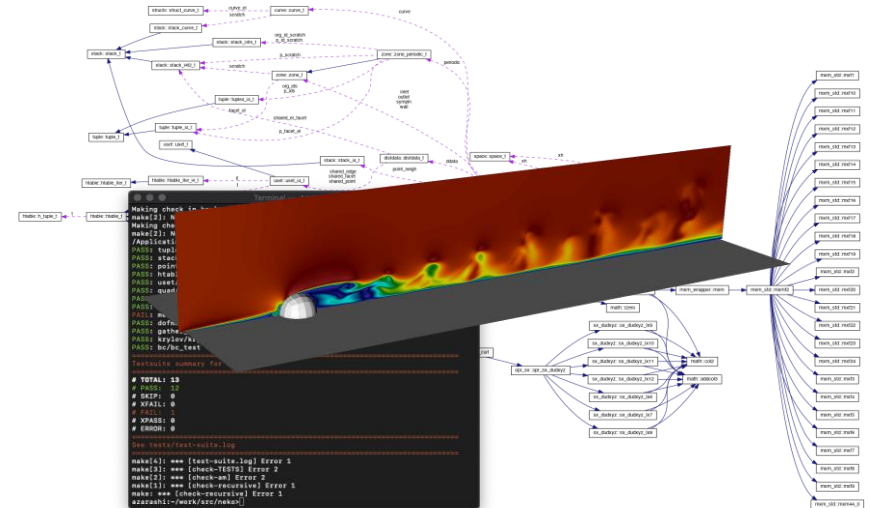
```

! Base type for a matrix-vector product providing Ax
type, abstract :: ax_t
contains
  procedure(ax_compute), nopass, deferred :: compute
end type ax_t

! Abstract interface for computing Ax
abstract interface
  subroutine ax_compute(w, u, coef, msh, Xh)
    implicit none
    type(space_t), intent(inout) :: Xh
    type(mesh_t), intent(inout) :: msh
    type(coef_t), intent(inout) :: coef
    real(kind=dp), intent(inout) :: w(:,:,:)
    real(kind=dp), intent(inout) :: u(:,:,:)
  end subroutine ax_compute
end interface

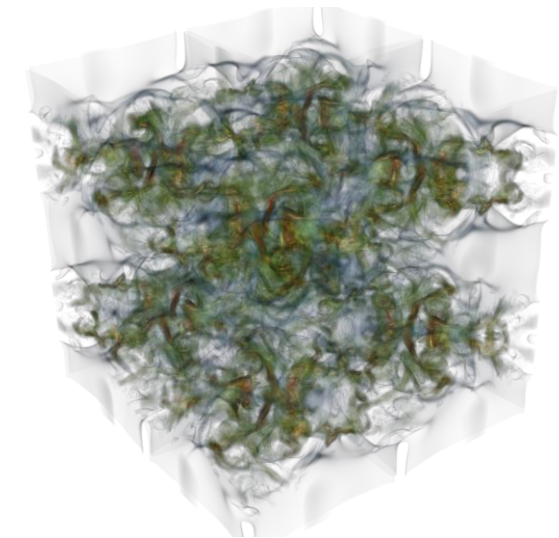
```

- Various hardware-backends
  - CPUs, GPUs down to exotic vector processors and FPGAs
    - > Regularly tested on Intel, AMD, A64FX and SX-Aurora
  - Multi-tier abstractions of the solver stack
    - > HW specific tuning e.g. kernel merging, redundant computation



N. Jansson et al. *Neko: A Modern, Portable, and Scalable Framework for High-Fidelity Computational Fluid Dynamics*, arXiv preprint arXiv:2107.01243, 2021.

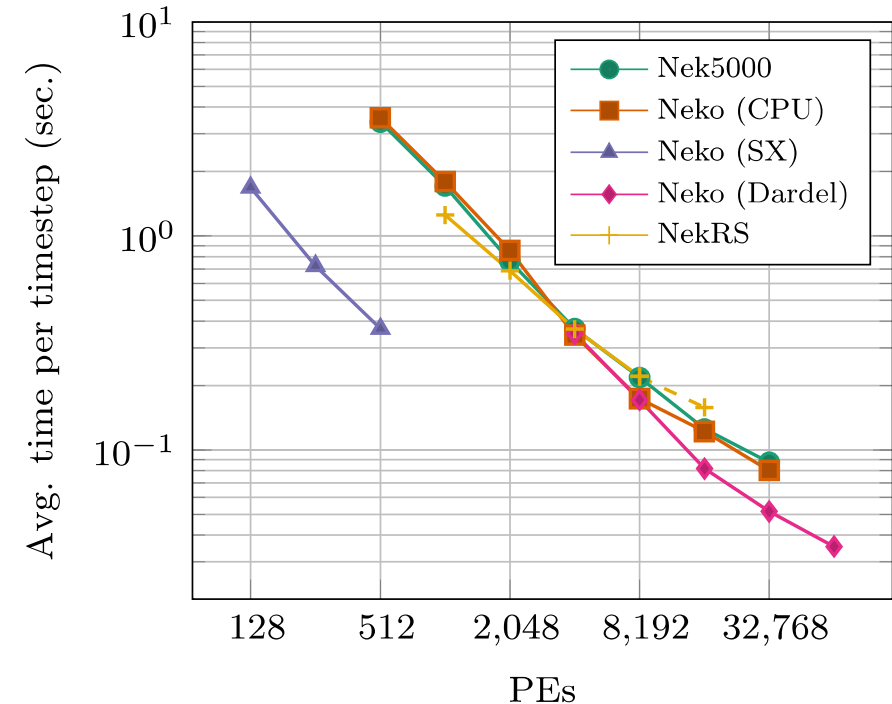
# Neko: Performance Evaluation



## Strong scalability test of Neko compared to NekRS/Nek5000

- Taylor-Green vortex at  $Re = 5000$ 
  - 261,144 hexahedral elements
  - 9<sup>th</sup> order polynomials (10 GLL points)
- **Beskow**, Cray XC40 at PDC
  - Two 16 cores Intel E5-2698v3
  - Tested on 16 up to 1024 nodes
- **Vulcan**, NEC A300-8 nodes at HLRS
  - Eight SX-Aurora Type 10B per node
  - Tested on 2 up to 8 nodes (entire machine)
- **Dardel**, HPE Cray EX at PDC
  - Two 64 cores AMD EPYC 7742
  - Tested on 32 up to 512 nodes

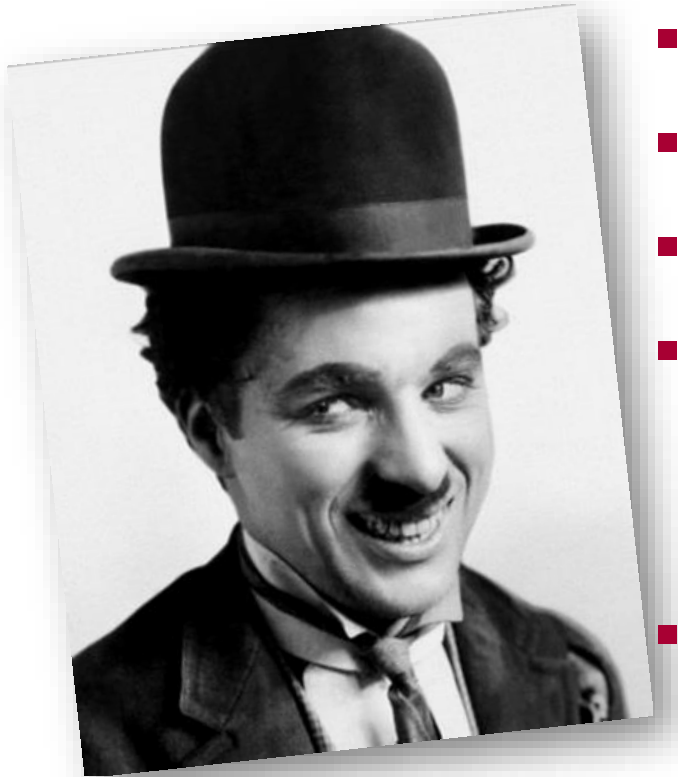
Taylor-Green vortex,  $Re = 5000$



<https://github.com/ExtremeFLOW/test-suite>

## To conclude...

- Let me put on a different hat...



- SNAC (Swedish National Allocation Committee)
- **Dardel phase 1 gives 47 Mh per month**
- For SNIC LARGE projects → 34 Mh/month
- 20 different large-scale projects, 7 universities in Sweden, 8 different areas of science
- National machine to the benefit of all HPC users!





# Acknowledgments

## KTH Engineering Mechanics

- Adam Peplinski
- Saleh Rezaeiravesh
- Adalberto Perez

## PDC

- Niclas Jansson
- Jonathan Vincent

## ENCCS

- Jing Gong